

SI No	DGCANO	SUBJECT	REFERENCE	COMPLIANCE	APPLICABILITY
1	DGCA/BOEING 747/1	CANCELLED IN VIEW OF FAA AD 71-06-04 R1	CANCELLED IN VIEW OF FAA AD NO. 71-06-04 R 1	AS IN AD	AS IN AD
2	DGCA/BOEING 747/2	INSPECTION FOR CRACKS AND REWORK OF NACELLE STRUT UPPER LINK.	F.A.A.A.D.NO. 71-06-07 AND BOEING S.B.NO. 747-54-2015REV.1	FIRST INSPECTION TO BE MADE AT THE NEXT TRIP INSPECTION REPEAT INSPECTION AS PER A.D.	AS IN A.D.
3	DGCA/BOEING 747/3	TO PREVENT MALFUNCTIONS RESULTING IN UN-USABLE PASSENGER EVACUATION SLIDE AND IMPROVE THE OVERALL RELIABILITY OF THE PASSENGER EVACUATION SYSTEM.	F.A.A.A.D.NO. 70-26-01 AND BOEING S.B.NO. 747-25-2031 REV.1	AS IN A.D.	AS IN A.D.
4	DGCA/BOEING 747/4	TIMKEN P/N LM229139 IN ALL LANDING GEAR WHEELS. ANY BEARING WITH CODE LETTER 'A' THROUGH 'F' MUST BE REPLACED WITH A BEARING WITH CODE LETTER 'G' OR LATER.	F.A.A.A.D.NO. 70-24-02	IMMEDIATE	AS IN A.D.
5	DGCA/BOEING 747/5	BODY GEAR STEERING ARM/DISARM SWITCH INSTALLATION.	F.A.A.A.D.NO. 71-17-02 AND BOEING S.B.NO.747-32-2113	AS IN A.D.	AS IN A.D.
6	DGCA/BOEING 747/6	STABILIZER TRIM CONTROL MODULAR PACKAGE INSPECTION AND REPLACEMENT.	F.A.A.A.D.NO. 72-04-01 AND BOEING S.B.NO.747-27-2054 REV.3	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
7	DGCA/BOEING 747/7	SHUTOFF FLUID PRESSURE REGULATING VALVE P/N 898108-2 -INSTALL BUTTERFLY STOP IN BODY ASSY.	ALERT S.B. AIRESEARCH FOR P/N 898108-2	AS IN S.B.	AS IN S.B.
8	DGCA/BOEING 747/8	INSPECTION OF WHEEL HUBS FOR CRACKS BY DYE PENETRANT METHOD.	F.A.A.A.D.NO.72-04-02	AS IN A.D.	AS IN A.D.
9	DGCA/BOEING/747/9	TO DETECT FAILED L/G ALTERNATE EXTENSION ACTUATORS.	F.A.A.A.D.NO.72-08-01	AS IN A.D.	AS IN A.D.
10	DGCA/BOEING 747/10	INSPECTION TO DETECT CRACK IN FORWARD ELECTRONIC FHATCH.	F.A.A.A.D.NO.74-08-10	AS IN A.D.	AS IN A.D.
11	DGCA/BOEING 747/11	FLAP CONTROL HANDLE INSPECTION AND REPLACEMENT.	F.A.A.A.D.NO.74-19-07 AND BOEING ALERT S.B.NO. 747-27-2121 REV.3	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
12	DGCA/BOEING 747/12	INSPECTION OF TORQUE TUBE CRANKS ON PASSENGER DOORS NOS. 2,4,5 RIGHT AND LEFT SIDES FOR CRACKS.	F.A.A.A.D.NO. 74-19-06	AS IN A.D.	AS IN A.D.
13	DGCA/BOEING 747/13	VISUAL INSPECTIOO OF ALL ELECTRICAL APPURTENANCES LOCATED WITHIN LAVATORY WASTE CONTAINER WERAR ABRASION AND CORROSION.	F.A.A.A.D.NO. 74-21-03	EFFECTIVE FROM 01/12/74	AS IN A.D.

14	DGCA/BOEING 747/14	FATIGUE CRACKS IN FUEL PUMP BODY P/N 968502-2 & 3 FITTED IN APU GTCP 660-4.	F.A.A.A.D.NO. 75-01-06 REV.	EFFECTIVE FROM 15-02-75	AS IN A.D.
15	DGCA/BOEING 747/15	INSTALLATION OF A PLACARD PROHIBITING INFLIGHT USE OF APU ON BOEING 747 AIRCRAFT AND A CHECK OF APU OPERATION TO BE PERFORMED FOLLOWING ANY AUTOMATIC/MANUAL SHUTDOWN.	F.A.A.A.D.NO. 75-05-17	AS IN A.D.	AS IN A.D.
16	DGCA/BOEING 747/16	PITOT STATIC SYSTEM TUBING AND DRAIN FITTING MODIFICATION.	F.A.A.A.D.NO. 75-04-06	AS IN A.D.	AS IN A.D.
17	DGCA/BOEING 747/17	AURAL/VISUAL INDICATION OF LEADING EDGE FLAP EXTENSION SYSTEM.	F.A.A.A.D.NO. 75-05-11	AS IN A.D.	AS IN A.D.
18	DGCA/BOEING 747/18	CONTROL CABLE PULLEYS IN THE SPEED BRAKE SYSTEM REPLACED WITH AVIONIC/ TAUSEY PULLEYS BACP 30(COLOUR BLACK ONLY).	F.A.A.A.D.NO. 75-05-01	AS IN A.D.	AS IN A.D.
19	DGCA/BOEING 747/19	INSTALLATION OF FOUR IDENTICAL VIBRATION ISOLATORS UNDER RACK SUPPORTING MODEL 209 DIGITAL FDR.	F.A.A.A.D.NO. 75-08-17	AS IN A.D.	AS IN A.D.
20	DGCA/BOEING 747/20	INFLIGHT DEPRESSURISATION CAUSED BY SUDDEN OPENING OF LARGE HOLE IN THE LOWER DECK CARGO COMPARTMENT.	F.A.A.A.D.NO. 75-15-05 REV.	ON OR BEFORE 31-12-78	AS IN A.D.
21	DGCA/BOEING 747/21	REMOVAL OF CAPACITOR ASSEMBLY FROM TOILET FUSH MOTOR AND CAPPING OF THE EXPOSED WIRES.	F.A.A.A.D.NO. 75-19-03	AS IN A.D.	AS IN A.D.
22	DGCA/BOEING 747/22	INSPECTION OF ALL FORE FLAP TRACK AIRLOAD ROLLERS, THE ATTATCH FITTINGS OF THE INBOARD TRAILING EDGE FORE FLAP AND THE OUTBOARD FORE FLAP INBOARD CARRIAGE FRAME.	F.A.A.A.D.NO. 75-20-05	AS IN A.D.	AS IN A.D.
23	DGCA/BOEING 747/23	INSPECTION/REPLACEMENT OF THE FLAP TRACKS.	F.A.A.A.D.NO. 76-03-06	AS IN A.D.	AS IN A.D.
24	DGCA/BOEING 747/24	FUSELAGE SKIN LAP INSPECTION UNDER WING TO BODY FAIRING.	F.A.A.A.D.NO. 76-05-05 AND BOEING S.B.NO. 747-53A2131 REV.3	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
25	DGCA/BOEING 747/25	REMOVAL OF UNAPPROVED APPLIANCES SUPPLIED BY OR PROCURED THROUGH ADS SUPPLYING COMPANY.	F.A.A.A.D.NO. 77-05-05	AS IN A.D.	AS IN A.D.
26	DGCA/BOEING 747/26	REPLACEMENT OF FUEL VALVE ACTUATOR ASSEMBLIES.	F.A.A.A.D.NO. 77-12-07	AS IN A.D.	AS IN A.D.
27	DGCA/BOEING 747/27	INSPECTION OF FORWARD AND AFT GATES OF EACH OUTFLOW VALVE ASSEMBLY TO PREVENT RAPID LOSS OF PRESURISATION.	F.A.A.A.D.NO. 78-08-04 AND BOEING S.B.NO. 747-21-2139 REV.1	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
28	DGCA/BOEING 747/28	CANCELLED IN VIEW OF DGCA/BOEING 747/30			
29	DGCA/BOEING 747/29	INSPECTION OF COOL GAS GENERTOR PROPELLENT CARTRIDGE.	F.A.A.A.D.NO. 78-05-08 REV.3 AND BOEING S.B.NO. 747-25-2373	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.

30	DGCA/BOEING 747/30	CANCELLED VIDE 9-456-80AI(2) DT. 07/83			
31	DGCA/BOEING 747/31	INSPECTION AND REWORK OF CARGO COMPARTMENT LINING.	F.A.A.A.D.NO. 78-07-09 AND BOEING S.B.NO. 747-25-2418	31-12-78 OR INCORPORATION OF MOD. AS PER A.D.75-15-05 WHICH EVER IS EARLIER	AS IN A.D. AND S.B.
32	DGCA/BOEING 747/32	TRAILING EDGE FLAP TRACK FORWARD FUSE BOLT INSPECTION AND REPLACEMENT.	F.A.A.A.D.NO. 83-08-02 AND BOEING S.B.NO. 747-57-2206 REV.1, 747-57-2217 REV.1.	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
33	DGCA/BOEING 747/33	INSPECTION,REWORK AND MODIFICATION OF MAIN FUEL TANK PUMP WIRING.	F.A.A.A.D.NO. 79-06-02 AND BOEING ALERT S.B.NO. 747-28A2092 REV.1	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
34	DGCA/BOEING 747/34	INSPECTION,REPAIR AND REPLACEMENT OF LOWER CARGO DOORWAY LOWER SILL TRUSS AND LATCH FITTING.	F.A.A.A.D.NO. 79-17-02 REV.2 AND BOEING S.B.NO. 747-53-2200 REV.1	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
35	DGCA/BOEING 747/35	INSPECTION OF PYLON ATTACH BOLTS FOR CRACKS.	F.A.A.A.D.NO.86-22-01 REV.1 AND BOEING S.B.NO. 747-54-2063 REV.5	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
36	DGCA/BOEING 747/36	INSPECTION OF INBOARD ENGINE STRUT TO DIAGONAL BRACE ATTACH FITTING.	F.A.A.A.D.NO. 95-20-05 AND BOEING S.B.NO.747-54-2062 REV.7	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
37	DGCA/BOEING 747/37	INSPECTION OF FIRING AND PACKBOARD LANYARDS INSTALLED ON ESCAPE SLIDE RAFTS ASSEMBLYS.	F.A.A.A.D.NO. 79-18-02 AND BOEING LAERT S.B.NO.747-25A2467 REV.3 AND GOODRICH ALERT S.B.NO. 25-051.	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
38	DGCA/BOEING 747/38	INSPECTION OF INBOARD PYLON DIAGONAL BRACE FORWARD FUSE PINS FOR CRACKS.	FAA AD 93-03-14 AND BOEING S.B.NO. 747-54-2066 REV.2	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
39	DGCA/BOEING 747/39	IMPROVED LIGHTENING PROTECTION DESIGN OF FUEL SYSTEM.	F.A.A.A.D.NO. 79-20-11	AS IN A.D.	AS IN A.D.
40	DGCA/BOEING 747/40	INSPECTION OF ENGINE STRUT FORWARD ENGINE MOUNT BULKHEAD STRUCTURE.	F.A.A.A.D.NO. 82-22-02 AND BOEING ALERT S.B.NO. 747-54A2069 REV.7	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.

41	DGCA/BOEING 747/41	INSEPTION OF OUTBOARD ENGINE STRUT DIAGONAL BRACE FORWARD CLEVIS ROOT.	F.A.A.A.D.NO. 80-08-02 AND BOEING S.B.NO. 747-54-2070 REV.2	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
42	DGCA/BOEING 747/42	TRAILING EDGE FLAP CARRIAGE SPINDLE CORROSION PROTECTION.	BOEING S.B.NO. 747-57-2133	01-10-80	AS IN S.B.
43	DGCA/BOEING 747/43	MODIFICATION OF LEADING EDGE FLAP CONTROL CIRCUIT.	F.A.A.A.D.NO. 80-19-05 AND BOEING S.B.NO. 747-27-2204 REV.3	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
44	DGCA/BOEING 747/44	REPLACEMENT OF FLAPPER VALVE IN B.F.GOODRICH SLIDES.	F.A.A.A.D.NO. 80-20-02 AND B.F.GOODRICH S.B.NO. 25-060	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
45	DGCA/BOEING 747/45	MODIFICATION OF STRINGERS 18 AND 19 BETWEEN FUSELAGE STN. 2360 AND 2412.	F.A.A.A.D.NO. 80-20-01 AND BOEING S.B.NO.747-53-2114 REV.4	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
46	DGCA/BOEING 747/46	REPLACEMENT OF MECHNANICAL RESTRAINTS ON ESCAPE SLIDES.	F.A.A.A.D.NO. 80-25-03 AND B.F.GOODRICH S.B.NOS. 25-054, 25-055, 25-056	02-01-84	AS IN A.D. AND S.B.
47	DGCA/BOEING 747/47	INSPECTION OF RACK PIN ASSEMBLY IN CREW COMPARTMENT OVERHEAD HATCH.	F.A.A.A.D.NO.81-15-06 AND BOEING S.B.NO. 747-52A2180 REV.1	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
48	DGCA/BOEING 747/48	FAIRING MOUNTED ESCAPE SLIDE FIRING MECHANISM RIGGING.	F.A.A.A.D.NO. 82-04-01 AND BOEING ALERT S.B.NO. 747-25A2581 REV.1	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
49	DGCA/BOEING 747/49	REWORK OF FAIRING MOUNTED ESCAPE SLIDE COOL GAS GENERATOR TRIGGER COMPONENTS.	BOEING S.B.NO. 747-25-2590	AS IN S.B.	AS IN S.B.
50	DGCA/BOEING 747/50	FLIGHT DECK OXYGEN HOSE/WIRE BUNDLE INSPECTION AND MODIFICATION.	BOEING ALERT S.B.NO. 747-35A2034	AS IN SB.	AS IN S.B.
51	DGCA/BOEING 747/51	INSPECTION/MODIFICATION OF MAIN DECK AND UPPER DECK LAVATORIES OXYGEN HOSE/WIRE BUNDLE INSTALLATIONS.	F.A.A.A.D.NO. 82-24-02 AND BOEING ALERT S.B.NO. 747-35A2032 REV.2	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
52	DGCA/BOEING 747/52	INSTALLATION OF PLACARD AND INCORPORATION OF LIMITATIONS IN THE FLIGHT MANUAL ON USE OF AUTOPILOT.	F.A.A.A.D.NO.82-26-01 AND BOEING S.B.NOS. 747-22-2127,747-22-2130	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.

53	DGCA/BOEING 747/53	MODIFICATION OF AIRCRAFT TO PREVENT ICING OF THE AILERON CONTROL CABLES.	F.A.A.A.D.NO. 83-02-10 AND BOEING S.B.NOS. 747-27-2065, 747-27-2095, 747-27-2161 AND 747-51-2032	AS IN A.D. AND S.B	AS IN A.D. AND S.B.
54	DGCA/BOEING 747/54	INSPECTION/REPLACEMENT OF WING LANDING GEAR JURY STRUT SPINDLES.	F.A.A.A.D.NO. 83-05-09 AND BOEING S.B.NO. 747-32-2261 REV.2	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
55	DGCA/BOEING 747/55	INSPECTION OF LONGITUDINAL FLOOR BEAMS AND REPAIR/MODIFICATION OF FLOOR BEAM WEBS AND CHORDS.	F.A.A.A.D.NO. 83-11-02 AND BOEING S.B.NOS. 747-53-2224 REV.6, 747-53-2176 REV.4 AND 747-53-2183 REV.2	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
56	DGCA/BOEING 747/56	INSTALLATION OF A PLACARD ON USE OF AUTOPILOT AND MODIFICATION OF LOW RANGE RADIO ALTIMETER.	F.A.A.A.D.NO. 83-13-08 AND BOEING S.B.NOS. 747-34A2225 AND 747-34A2226	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
57	DGCA/BOEING 747/57	ENGINE AND LOWER CARGO FIRE EXTINGUISHER OF BOTTLE DISCHARGE SWITCH INSPECTION AND LUBRICATION/REPLACEMENT.	BOEING ALERT S.B.NO. 747-26A2098 REV.2	AS IN S.B.	AS IN S.B.
58	DGCA/BOEING 747/58	INBOARD AND OUTBOARD FLAP TRACK DRAIN HOLE ADDITION AND CORROSION INSPECTION.	F.A.A.A.D.NO. 83-21-02 REV.1 AND BOEING S.B.NO. 747-57A2225 REV.4.	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
59	DGCA/BOEING 747/59	INSPECTION AND REWORK OF EMERGENCY DESCENT DEVICES.	BOEING ALERT S.B.NO. 747-25A2627	AS IN S.B.	AS IN S.B.
60	DGCA/BOEING 747/60	INSPECTION AND/OR REPLACEMENT OF INBOARD STRUT DIAGONAL BRACE TO WING FUSE PIN.	FAA AD 93-03-14 AND BOEING S.B.NO. 747-54-2101	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
61	DGCA/BOEING 747/61	AMENDMENT TO AFM AND INSTALLATION OF LOW N1 CONDITION INDICATOR.	F.A.A.A.D.NO. 84-02-05	AS IN A.D.	AS IN A.D.
62	DGCA/BOEING 747/62	REPLACEMENT OF LANDING GEAR COMPONENTS.	F.A.A.A.D.NO. 84-03-06	AS IN A.D.	AS IN A.D.
63	DGCA/BOEING 747/63	SUPPLEMENTAL STRUCTURAL INSPECTION PROGRAMME FOR BOEING 747 AIRCRAFT.	FAA AD 93-06-01 AND BOEING SSID DOCUMENT NO. D6-35022R4 DT.FEB'1992	AS IN A.D. AND DOC.	AS IN A.D. AND DOC.

64	DGCA/BOEING 747/64	FUSELAGE - KEEL BEAM BOX - KEEL BEAM BODY STN.1265 INTERNAL DIAPHRAGM BACK UP BRACKET INSPECTION.	BOEING ALERT S.B.NO. 747-53A2245	AS IN S.B	AS IN S.B
65	DGCA/BOEING 747/65	INSPECTION OF BODY STATION 1241 BULKHEAD SPLICE STRAP FOR CRACKS.	FAA AD 92-08-02	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
66	DGCA/BOEING 747/66	INSPECTION OF NOSE GEAR WHEEL WELL LOWER FORWARD COVERS EXTERIOR AND INTERIOR AREA FOR CRACKS.	F.A.A.D.NO. 89-04-05 AND BOEING S.B.NO. 747-53-2112 REV.3.	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
67	DGCA/BOEING74 7/67(R1)	TO DETECT AND CORRECT FATIGUE CRACKS IN THE BODY STATION 1000 BULKHEAD CHORD	FAA AD 2005-10-21 (AD 90-09-09 IS SUPERSEDED)	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
68	DGCA/BOEING 747/68	FRAME TO TENSION TIE JOINT MODIFICATION - BODY STATION 760 OR 780.	F.A.A.D.NO. 84-19-01 AND BOEING S.B.NO. 747-53-2088 REV.2.	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
69	DGCA/BOEING 747/69	INSPECTION FLAP TRACK LOWER FLANGES AND VERTICAL WEBS AT THE FRONT END.	F.A.A.D.NO. 89-05-04 R1 AND BOEING S.B.NO. 747-57A2229 REV.7.	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
70	DGCA/BOEING 747/70	FLIGHT CONTROLS ELEVATOR - INBOARD ELEVATOR CONTROL RODS AND AFT QUADRANT INSPECTION AND REPLACEMENT.	FAA AD 90-25-10 AND BOEING ASB 747-27A2253 REV.4	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
71	DGCA/BOEING 747/71	INSTALLATION OF A PLACARD 'DO NOT USE THE LAND MODE' ON THE AUTO PILOT P 10 MODE SELECT PANEL.	FAA AD 84-25-04	AS IN AD	AS IN AD
72	DGCA/BOEING 747/72	REPLACEMENT OF PRESSURE REGULATOR 'O' RING ON AIRPLANES EQUIPPED WITH B.F. GOOD RICH SLIDES OR SLIDENAFFS	FAA, AD, 85-02-01	AS IN AD	AS IN AD
73	DGCA/BOEING 747/73	TO PREVENT FAILURE OF THE FUSELAGE SKIN LAP SLIDE BETWEEN BODY STS. 340 & 400 AT STRINGESS S6-L & S6-R.	FAA AD 90-23-14	AS IN AD	AS IN AD
74	DGCA/BOEING 747/74	INSPECTION OF ALL PRESSURE BULKHEAD FOR EVIDENCE OF REPAIR OR DAMAGE	F.A.A. AD 85-22-12	AS IN AD	AS IN AD
75	DGCA/BOEING 747/75	ULTRASONIC INSPECTION OF THE INBOARD AND OUTBOARD MIDSPAN FITTING LUGS	FAA AD 93-17-07 AND SB 747-54-2100 AND SB 747-54A-2152	AS IN AD AND SBS	AS IN AD AND SBS
76	DGCA/BOEING 747/76(R1)	INSPECTION OF BODY FRAME STRUCTURE AND SKIN FOR CRACKING	FAA AD 2005-08-01 (ORIGINAL AD 91-11-01 IS SUPERSEDED)	AS IN AD	AS IN AD
77	DGCA/BOEING 747/77	INSPECTION AND REPAIR OF NO. 1 ENGINE CONTROL CABLE.	FAA AD T-85-25-55 AND BOEING ALERT SB 747-76A-2063	AS IN AD	AS IN AD

78	DGCA/BOEING 747/78	NACELLE STRUT ATTACH FITTINGS - INBOARD AND OUTBOARD FITTING FASTENER INSPECTION AND REPLACEMENT.	F.A.A.A.D.NOS. 86-23-01,86-05-11 REV.1 AND BOEING ALERT S.B.NOS. 747-57A2233 REV.2,747-57A2235 REV.3.	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
79	DGCA/BOEING 747/79	DOOR MOUNTED ESCAPE SLIDE/RAFT PACKBOARD STRAP ADDITION.	BOEING S.B.NO. 747-25A2680 REV.1	AS IN S.B.	AS IN S.B.
80	DGCA/BOEING 747/80	VERTICAL FIN ACCESS COVER INSTALLATION IN BODY FIN DECK OPENING.	F.A.A.A.D.NO. 86-08-02 AND BOEING S.B.NO. 747-53A2264	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
81	DGCA/BOEING 747/81	INSPECTION OF FORWARD LUG OF INBOARD PYLON UPPER LINK FOR CRACKS.	F.A.A.A.D.NO. 86-07-06 AND BOEING S.B.NO. 747-54-2111 REV.3	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
82	DGCA/BOEING 747/82	EXTERNAL VISUAL INSPECTION OF LOWER LONGITUDINAL LAP JOINT AREAS FOR CORROSION.	F.A.A.A.D.NO. 86-09-07 REV.1 AND BOEING S.B.NO. 747-53A2267 REV.1 DT.25TH SEP.'86	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
83	DGCA/BOEING 747/83R1	TO DETECT & CORRECT THE LOSS OF STRUCTURAL INTEGRITY OF THE FUSELAGE ,WHICH COULD RESULT IN RAPID DEPRESSURIZATION OF THE AIRPLANE.	FAA AD 2009-25-11	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
84	DGCA/BOEING 747/84	INSPECTION AND MODIFICATION OF AFT CARGO BAY FRAME - SECTION 46.	BOEING ALERT S.B.NO. 747-53A2259 REV.1	AS IN S.B.	AS IN S.B.
85	DGCA/BOEING 747/85	TO PREVENT FAILURE OF THE ENGINE THRUST CONTROL CABLE AND LOSS OF CONTROL OF AN ENGINE.	F.A.A.A.D.NO. 86-10-10 AND BOEING S.B.NO. 747-76A2065 REV.2	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
86	DGCA/BOEING 747/86	TO PREVENT WHEEL OR TYRE FAILURE AS A RESULT OF FAILURE OF THE WHEEL BEARING.	F.A.A.A.D.NO. 86-11-01 REV.1	AS IN A.D.	AS IN A.D.
87	DGCA/BOEING 747/87	EQUIPMENT FURNISHINGS - EMERGENCY DOOR MOUNTED ESCAPE SLIDE COOL GAS GENERATOR INFLATION CABLE NUT INSPECTION.	F.A.A.A.D.NO. 86-16-06 AND BOEING ALERT S.B.NO. 747-25A2696 REV.2	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
88	DGCA/BOEING 747/88	WINGS - TRAILING EDGE FLAPS - FLAP TRACK NOS. 1,2,3,6,7 & 8 FORWARD ATTACHMENT FASTENERS INSPECTION AND REPLACEMENT.	F.A.A.A.D.NO. 86-18-06 AND BOEING ALERT S.B.NO. 747-57A2234 REV.1	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
89	DGCA/BOEING 747/89	TO DETECT & CORRECT FATIGUE CRACKING OF THE BULKHEAD WEB	F.A.A.A.D.NOS.2000-15-08.	AS IN A.D AND S.B.	AS IN A.D AND S.B.

90	DGCA/BOEING 747/90	EQUIPMENT/FURNISHINGS - DOOR MOUNTED ESCAPE SLIDE RAMP PACK ESCAPE SLIDE SUPPORT BRACKETS REVISION AND MODIFICATION.	F.A.A.A.D.NO. 87-01-02 AND BOEING ALERT S.B.NOs. 747-25A2710 REV.1 AND 747-25A2747 REV.1	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
91	DGCA/BOEING 747/91	TO PRECLUDE THE POTENTIAL FOR SEPARATION OF THE FIRE EXTINGUISHER DISCHARGE OUTLET AND MISDIRECTION OF THE FIRE EXTINGUISHING AGENT.	F.A.A.A.D.NO. 86-19-03 AND BOEING S.B.NOS.747-26A2108 REV.2,7 47-26A2110 REV.2	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
92	DGCA/BOEING 747/92	DOORS - PASSENGER/CREW - MAIN ENTRY DOOR EMERGENCY POWER SYSTEM - EMERGENCY POWER RESERVOIR SERIAL NO. INSPECTION/REWORK.	F.A.A.A.D.NO. 87-07-06 AND BOEING ALERT S.B.NO. 747-52A2201 REV.1	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
93	DGCA/BOEING 747/93	STABILIZERS - HORIZONTAL STABILIZER - FORWARD AND REAR SPAR CHORD INSPECTION.	BOEING ALERT S.B.NO. 747-53A2028 DT.15-01-87	AS IN S.B.	AS IN S.B.
94	DGCA/BOEING 747/94	FUSELAGE - SKIN - UPPER BODY SKIN BONDED TEAR STRAP INSPECTION AND MODIFICATION.	F.A.A.A.D.NO. 87-16-13 AND BOEING ALERT S.B.NO. 747-53A2279 REV.1	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
95	DGCA/BOEING 747/95	INSPECTION AND MODIFICATION OF EMERGENCY ESCAPE SYSTEM	F.A.A.A.D.NO. 87-04-07 AND BOEING S.L.NO. 747-SL-52-35 REV.A	AS IN A.D.	AS IN A.D.
96	DGCA/BOEING 747/96	INSPECTION FOR CRACKING AND REPAIR OR REPLACEMENT OF THE PYLON MIDSPAR ATTACH FITTING.	F.A.A.A.D.NO. 97-12-03.	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
97	DGCA/BOEING 747/97	INSPECTION AND REPLACEMENT OF STATION 2598 LONGERON SKIN SPLICE FITTINGS AND TENSION BOLTS.	F.A.A.A.D.NO. 87-24-02 AND BOEING S.B.NO.747-53A2280 REV.1	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
98	DGCA/BOEING 747/98	INSTALLATION OF HYDRAULIC FUSE IN NO.4 HYDRAULIC SYSTEM.	F.A.A.A.D.NO. 87-12-04 AND BOEING S.B.NO. 747-29-2063 REV.3	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
99	DGCA/BOEING 747/99	FORWARD AND AFT CARGO DOORS - LATCH LOCKING SYSTEM TESTS, OPERATION AND MODIFICATION.	FAA AD NO.90-09-06 AND BOEING S.B.NOS. 747-52A-2206 REV.5 AND 747-52A- 2209 REV.2	AS IN A.D. & S.B.	AS IN A.D. & S.B.

100	DGCA/BOEING 747/100	TO DETECT & CORRECT FAILED CARRIAGE SPINDLES OR AFT LINKS OF THE INBOARD OR OUTWARD TRAILING EDGE FLAPS.	FAA AD 2008 - 22 - 17	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
101	DGCA/BOEING 747/101	INSPECTION OF WING LEADING EDGE PNEUMATIC DUCT.	FAA AD 91-06-08	AS IN AD	AS IN AD
102	DGCA/BOEING 747/102	REPLACEMENT OF EMERGENCY DOOR DEPLOYMENT THRUSTER 'O' RINGS.	FAA AD 97-03-19.	AS IN AD AND S.B.	AS IN AD AND S.B.
103	DGCA/BOEING 747/103	INSPECTION AND REPLACEMENT OF APU FUEL SUPPLY LINES.	F.A.A. A.D.NO. 88-05-07 AND BOEING S.B.NO. 747-49-2051 REV.1	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
104	DGCA/BOEING 747/104	INSPECTION FOR FUEL LEAKS FROM CENTRE WING FUEL TANK FRONT SPAR INTO THE FORWARD CARGO COMPARTMENT.	FAA AD 90-11-08 AND BOEING ASB 747-57A2247 REV.3	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
105	DGCA/BOEING 747/105	INSPECTION,REPAIR AND MODIFICATION OF SKIN LAP SPLICE ALONG STRINGER 6 LEFT AND RIGHT FROM BODY STATION 400 TO 520.	FAA AD 90-21-17	AS IN AD	AS IN AD
106	DGCA/BOEING 747/106	INSPECTION OF THE THRUST CONTROL CABLE BRACKET ASSEMBLY LOCATED AT ILES 1116 FOR PROPER INSTALLATION.	FAA A.D. NO.89-08-09 & BOEING ALERT S.B.NO. 747-76A2074 REV.1	AS IN A.D. & S.B.	AS IN A.D. & S.B.
107	DGCA/BOEING 747/107	MODIFICATION OF MAIN LANDING GEAR DOWNLOCK ACTUATOR.	BOEING ALERT S.B. NO. 747-32A2320 REV. 2.	AS IN S.B.	AS IN S.B.
108	DGCA/BOEING 747/108	INSPECTION OF ENGINE STRUT DIAGONAL BRACE LUGS FOR CRACKS.	FAA A.D. NO. 89-07-15 & BOEING S.B. NO. 747-54-2126	AS IN A.D. & S.B.	AS IN A.D. & S.B.
109	DGCA/BOEING 747/109	INSPECTION OF FUSELAGE BETWEEN BS 940 AND BS 1000 BODY CROWN CREASE BEAM AND THE INTERCOSTAL STRUCTURE FOR CRACKS.	FAA A.D. NO.89-08-03 REV. 1 & BOEING S.B. NO.747-53-2297	AS IN A.D.	AS IN A.D.
110	DGCA/BOEING 747/110	REPLACEMENT OF UNDERWING FUEL TANK ACCESS DOORS.	F.A.A. A.D.NO. 89-12-10 BOEING ALERT S.B.NO. 747-28A2117 REV. 2	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
111	DGCA/BOEING 747/111	MODIFICATION OF OUT BOARD STRUT TRAILING EDGE FAIRING DOORS SEAL.	FAA A.D. 89-16-07 AND BOEING ALERT S.B.NO.747-54A2129	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
112	DGCA/BOEING 747/112	INSPECTION AND CLEANING OF THE CAVITY AFT OF THE WING CENTRE SECTION.	F.A.A. A.D.NO. 89-12-07	AS IN A.D.	AS IN A.D.

113	DGCA/BOEING 747/113	MODIFICATION OF DOOR MOUNTED ESCAPE SLIDE RAFT LOWER LINING.	BOEING ALERT S.B.NO. 747-25A2803	AS IN S.B.	AS IN S.B.
114	DGCA/BOEING 747/114	REPLACEMENT OF TAKEOFF WARNING SYSTEM SWITCH ASSEMBLY MOUNTING BRACKETS.	FAA A.D. NO. 89-15-02	AS IN A.D.	AS IN A.D.
115	DGCA/BOEING 747/115	INSPECTION OF THE WING LANDING GEAR BEAM OUTBOARD END FITTINGS FOR CRACK AND CORROSION.	FAA A.D. 89-15-07 AND BOEING S.B. 747-57-2244 REV.1	AS IN AD AND SB.	AS IN AD AND SB
116	DGCA/BOEING 747/116	INSTALLATION OF PROTECTIVE SLEEVES ON WIRE BREAKOUTS AND OXYGEN LINES IN THE UPPER DECK LEFT SIDEWALL.	FAA AD 91-13-11 & BOEING SB 747-35-2059 R2	AS IN AD & SB	AS IN AD & SB
117	DGCA/BOEING 747/117	VISUAL AND EXTERNAL HF EDDY CURRENT INSPECTION FOR CRACKS BETWEEN BS 220 TO BS 520 LEFT AND RIGHT SIDE OF AIRPLANE AND BETWEEN S6' AND S20.	FAA AD T89-23-53 & BOEING ALERT SB 747-53-A2321	AS IN AD & SB	AS IN AD & SB
118	DGCA/BOEING 747/118	INSPECTION, REPAIR, MODIFICATION OF FUSELAGE SKIN AT LEFT AND RIGHT BS 460 CIRCUMFERENTIAL SPLICES BETWEEN STRINGERS S-14E AND S-16.	FAA AD 89-21-09 & BOEING SB 747-53-2289 REV. 1	AS IN AD & SB	AS IN AD & SB
119	DGCA/BOEING 747/119	DOORS- PASSENGER/ CREW- MAIN ENTRY DOORS- MAIN ENTRY DOOR 3 EMERGENCY LINKAGE- ROD END BEARING INSPECTION.	FAA AD 90-10-03 & BOEING ALERT SB 747-52A2217	AS IN AD & ASB	AS IN AD & ASB
120	DGCA/BOEING 747/120	TO MINIMISE TH HAZARD ASSOCIATED WITH MAIN DECK CLASS B CARGO COMPARTMENT FIRE.	FAA AD 93-07-15	AS IN AD	AS IN AD
121	DGCA/BOEING 747/121	VISUAL INSPECTION OF H-11 STEEL BOLTS FOR CRACKS OR FRACTURE AND REPLACEMENT.	FAA AD 89-23-07 & BOEING SB 747-51-2043	AS IN AD & SB	AS IN AD & SB
122	DGCA/BOEING 747/122	INSPECTION OF HORIZONTAL STABILIZER HINGE FITTING BACKUP STRUCTURE FASTENERS.	FAA AD 89-23-19 & BOEING ASB 747-53A2314	AS IN AD & SB	AS IN AD & SB
123	DGCA/BOEING 747/123	FUSELAGE-TAIL SECTION 2360-2792-APU FIREWALL PLUG ASSEMBLY REPLACEMENT.	FAA AD 90-11-02 & BOEING ASB 53A2317	AS IN AD & SB	AS IN AD & SB
124	DGCA/BOEING 747/124	TO PREVENT REDUCED STRUCTURAL INTEGRITY OF THE FUESLAGE.	FAA AD 2008-16-14 SUPERSEDES FAA AD 94-15-06 & BOEING ASB 747-53A2312 LATEST REV.	AS IN AD & ASB	AS IN AD & ASB
125	DGCA/BOEING 747/125	STRUCTURAL MODIFICATION- AGING AIRPLANES	FAA AD 90-06-06	AS IN AD	AS IN AD
126	DGCA/BOEING 747/126	INSPECTION OF WING LANDING GEAR AFT TRUNNION FOR CRACKS AND CORROSION.	FAA AD 2001-17-25 & BOEING SB 747-32-2190 R4	AS IN AD & SB	AS IN AD & SB
127	DGCA/BOEING 747/127	INSPECTION/ REPAIR OF CRACKS IN THE FUSELAGE STATION 2598 UPPER BULKHEAD.	FAA AD 91-05-17 & BOEING ASB 747-53A2332	AS IN AD & SB	AS IN AD & SB
128	DGCA/BOEING 747/128	MODIFICATION OF BFGOODRICH DOOR 3 LOWER DOOR COVER.	FAA AD 90-13-09 AND BF GOODRICH ASB 25-229	AS IN AD AND SB	AS IN AD AND SB

129	DGCA/BOEING 747/129	MODIFICATION OF BFGOODRICH DOOR 3 LOWER DOOR COVER.	FAA AD 90-13-09 AND BF GOODRICH ASB 25-229	AS IN AD AND SB	AS IN AD AND SB
130	DGCA/BOEING 747/130	VISUAL INSPECTION OF NOSE WHEEL VERTICAL BEAMS.	FAA AD T90-15-51 & BOEING ASB 747-53A2338	AS IN AD & SB	AS IN AD & SB
131	DGCA/BOEING 747/131	NACELLES/PYLONS-NACELLE STRUT FUSE PIN-INBOARD AND OUTBOARD UPPER LINK, OUTBOARD DIAGNOL BRACE FUSE PIN REPLACEMENT.	BOEING SB 747-54-2102	AS IN SB	AS IN SB
132	DGCA/BOEING 747/132	INSPECTION OF THE NOSE WHEEL WELL BOX STRUCTURE FOR CRACKS.	FAA AD 90-06-14 R1 & FAA AD 90-15-51.	AS IN AD.	AS IN AD.
133	DGCA/BOEING 747/133R1	AIR TRANSPORT ASSOCIATION (ATA) OF AMERICA CODE 53:FUSELAGE	FAA AD 2010-21-04	AS IN AD & SB.	AS IN AD & SB.
134	DGCA/BOEING 747/134	INSPECTION OF THE NACELLE STRUT DIAGONAL BRACE FOR CRACKING.	FAA AD 90-20-20	AS IN AD	AS IN AD
135	DGCA/BOEING 747/135	INSPECTION OF THE NACELLE STRUT DIAGONAL BRACE FOR CRACKING.	FAA AD 90-20-20	AS IN AD.	AS IN AD.
136	DGCA/BOEING 747/136	REPLACEMENT OF THE TRAILING EDGE FLAP TRACK FAIL-SAFE STRAPS/ BOLTS.	FAA AD 90-20-10	AS IN AD	AS IN AD
137	DGCA/BOEING 747/137	INSPECTION/ TESTING/ REPAIR/ REPLACEMENT OF ENGINE BLEED AIR CROSSOVER DUCTS.	FAA AD 90-21-09	AS IN AD	AS IN AD
138	DGCA/BOEING 747/138	MODIFICATION OF CERTAIN HYDRAULIC LINES IN THE NUMBER 2 AND 3 STRUTS.	FAA AD 90-21-13	AS IN AD	AS IN AD
139	DGCA/BOEING 747/139	INSPECTION OF FLAP TRACK FORWARD END CLEVIS LUG.	FAA AD 2003-14-15	AS IN AD	AS IN AD
140	DGCA/BOEING 747/140	AGING AIRPLANE CORROSION PREVENTION AND CONTROL PROGRAM.	FAA AD 90-25-05R1 REVISES FAA AD 90-25-05 AND BOEING DOC. NO. D6-36022 REV. A	AS IN AD AND DOC.	AS IN AD AND DOC.
141	DGCA/BOEING 747/141	INSTALLATION OF ADDITIONAL FIRE RESISTANT CARGO COMPARTMENT LINER PANELS.	FAA AD 90-25-08	AS IN AD	AS IN AD
142	DGCA/BOEING 747/142	TO VERIFY PROPER APPLICATION OF THE CENTRE WING FUEL TANK SECONDARY FUEL BARRIER AND PREVENT FUEL OR FUEL VAPOURS FROM ENTERING THE PASSENGER AND CARGO COMPARTMENT.	FAA AD 91-03-13	AS IN AD	AS IN AD
143	DGCA/BOEING 747/143	INSPECTION OF THE TRAILING EDGE FLAP TRACKS FAILSAFE BAR BOLT HOLES FOR CORROSION, CRACK AND ADEQUATE CADMIUM PLATING.	FAA AD 2002-23-21	AS IN AD	AS IN AD
144	DGCA/BOEING 747/144	INSPECTION OF EACH OUTBOARD ENGINE FUEL FEED LINE, REPAIR/ REPLACEMENT.	FAA AD 96-11-03.	AS IN AD	AS IN AD
145	DGCA/BOEING 747/145	INSPECTION FOR CRACKS OF THE SECTION 42 FRAMES.	FAA AD 91-06-03	AS IN AD	AS IN AD

146	DGCA/BOEING 747/146	INSPECTION FOR DAMAGE TO THE ENGINE FIRE EXTINGUISHING TUBES.	FAA AD 91-08-06	AS IN AD	AS IN AD
147	DGCA/BOEING 747/147	INSPECTION OF NACELLE STRUT MIDSPAR FUSE PINS	FAA AD 91-09-01	AS IN AD	AS IN AD
148	DGCA/BOEING 747/148	INSTALLATION OF SHIELDS TO PREVENT ENTRY OF UNWANTED MATERIALS INTO THE MID TO AFT DRAIN MAST ASSEMBLIES.	FAA AD 91-09-11	AS IN AD	AS IN AD
149	DGCA/BOEING 747/149	PENETRANT INSPECTION AND PROOF TESTING OF THE APU PNEUMATIC DUCTS	FAA AD 91-15-24	AS IN AD	AS IN AD
150	DGCA/BOEING 747/150	INSPECTION OS FUSELAGE LOWER FRAMES AT BS2200 AND BS2220	FAA AD 91-18-04	AS IN AD	AS IN AD
151	DGCA/BOEING 747/151	REPETITIVE INSPECTION OF THE ENGINE NO.2 AND ENGINE NO. 3 UPPER STRUT WING L.E COMPARTMENTS FOR CHAFING AND OTHER DAMAGE TO THE FUEL SUPPLY TUBE AND ELECTRICAL POWER FEEDER CABLES.	FAA AD T91-20-51	AS IN AD	AS IN AD
152	DGCA/BOEING 747/152	INSPECTION OF LANDING GEAR BRAKE WEAR LIMITS INTO THE FAA APPROVED MAINTENANCE PROGRAM.	FAA AD 92-18-12	AS IN AD	AS IN AD
153	DGCA/BOEING 747/153	INSTALLATION OF A MODIFIED EVACATION RAMP SLIDE REGULATOR ASSEMBLY.	FAA AD 91-25-02	AS IN AD	AS IN AD
154	DGCA/BOEING 747/154	EXAMINATION OF FLOW CONTROL UNITS FOR THE PASSENGER OXYGEN SYSTEM AND REPLACEMENT OF CERTAIN UNITS.	FAA AD 92-04-03	AS IN AD	AS IN AD
155	DGCA/BOEING 747/155	INSTALLATION OF APU PNEUMATIC DUCT RESTRAINT	FAA AD 92-04-05	AS IN AD	AS IN AD
156	DGCA/BOEING 747/156	INSTALLATION OF A DECOMPRESSION PANEL AND FLAPPER VALVE IN THE AFT LOWER LOBE AT STATION 1920	FAA AD 92-02-13	AS IN AD	AS IN AD
157	DGCA/BOEING 747/157	INSPECTION AND MODIFICATION OF THE EMERGENCY ESCAPE SYSTEM PACKBOARD ASSEMBLIES.	FAA AD 92-06-03	AS IN AD	AS IN AD
158	DGCA/BOEING 747/158	ONE TIME INSPECTION, TEST AND REPAIR OR REPLACEMENT, IF NECESSARY, OF THE GUIDE ARM ASSEMBLY OF THE MAIN ENTRY DOORS NOS.1 THROUGH 5, LEFT AND RIGHT SIDES.	FAA AD 92-06-07	AS IN AD	AS IN AD
159	DGCA/BOEING 747/159	INSPECTION AND MODIFIACION OF THE LIFE RAFT MOORING LINE AND INFLATION LENGTH.	FAA AD 92-14-02	AS IN AD	AS IN AD
160	DGCA/BOEING 747/160	REPLACEMENT OF THE PLASTIC PULLEY GUARDS ON EMERGENCY ESCAPE SLIDE SYSTEM.	FAA AD 92-16-08	AS IN AD	AS IN AD
161	DGCA/BOEING 747/161	INSPECTION / REPLACEMENT OF NACELLE STRUT MIDSPAR FUSE PINS	FAA AD 92-24-51 AND SB 747-54-2063 REV.9	AS IN AD AND SB	AS IN AD AND SB
162	DGCA/BOEING 747/162	REPLACEMENT OF CERTAIN SECTIONS OF THE ENGINE FUEL SUPPLY LINE IN THE NUMBER ONE, NO. 3, AND NO. 4 ENGINE STRUTS.	FAA AD 92-06-01 AND SB 747-28-2153	AS IN AD AND SB	AS IN AD AND SB
163	DGCA/BOEING 747/163	INSPECTION/REPAIR TO CERTAIN UPPER DECK FLOOR BEAMS.	FAA AD 92-24-07 AND SB 747-53-2327	AS IN AD AND SB	AS IN AD AND SB

164	DGCA/BOEING 747/164	STRUCTURAL INSPECTIONS OF OLDER AIRPLANES.	FAA AD 92-27-04 AND AGING AIRPLANE SB STRUCTURAL MODIFICATION AND INSPECTION PROGRAM, SECTION 4 OF BOEING DOCUMENT NO. D6-35999 REV.C	AS IN AD AND SB	AS IN AD AND SB
165	DGCA/BOEING 747/165	VISUAL INSPECTION OF THE WIRE BUNDLES BETWEEN P6 AND P7 PANELS, 6" AFT OF THE P6 PANEL	FAA AD 92-27-12 AND BOEING STANDRAD WIRING DOCUMENT D6-54446	AS IN AD AND DOCUMENT	AS IN AD AND DOCUMENT
166	DGCA/BOEING 747/166	REPETITIVE HIGH FREQUENCY EDDY CURRENT INSPECTION OF LATCH SUPPORT FITTING OF CARGO DOOR FOR CRACKS.	FAA AD 98-18-23	AS IN AD AND SB	AS IN AD AND SB
167	DGCA/BOEING 747/167	MODIFICATION OF EMERGENCY DOOR MOUNTED ESCAPE SLIDE GIRT BAR FITTINGS.	FAA AD 92-02-17 AND BOEING SB 747-25-2754 R1	AS IN AD AND SB	AS IN AD AND SB
168	DGCA/BOEING 747/168	INSPECTION/ REWORK/ REPLACEMENT OF DAMAGED PARTS OF CERTAIN LATCH PINS SUPPORT FITTING INSTALLATIONS.	FAA AD 93-10-01 AND BOEING SB 747-52-2186 R4 AND ASB 747-52A-2233	AS IN AD AND SB	AS IN AD AND SB
169	DGCA/BOEING 747/169	EXAMINATION OF LOCKING GROOVES OF THE LATCHING HANDLE, AND ITS REPLACEMENT/ ADJUSTMENT.	FAA AD 91-22-05 AND BOEING SB 747-25-3013	AS IN AD AND SB	AS IN AD AND SB
170	DGCA/BOEING74 7/170	TO PREVENT THE LOSS OF THE STRUCTURAL INTEGRITY OF THE FUSELAGE WHICH COULD RESULT IN RAPID DEPRESSRIZATION OF THE AIRPLANE.	FAA AD 2009 -18 -07	AS IN AD AND SB	AS IN AD AND SB
171	DGCA/BOEING 747/171	INSEPTION OF THE FIRE EXTINGUISHING DISCHARGE TUBES AND SUPPORT CLAMPS IN THE OUTBOARD ENGINES - REPAIR OR REPLACEMENT, IF NECESSARY.	FAA AD 93-18-07 AND BOEING ASB 747-26A2214	AS IN AD AND SB	AS IN AD AND SB
172	DGCA/BOEING 747/172	MODIFICATION OF BF GODRICH SLIDE RAFTS.	FAA AD 93-20-04 AND BF GODRICH SB 4A3221-25-250 DT. MARCH 12 '1993	AS IN AD AND SB.	AS IN AD AND SB.
173	DGCA/BOEING 747/173	REVISION TO THE FAA APPROVED MAINTENANCE PROGRAM TO REQUIRE UNRESTRICTED OPENING OF THE PASSENGER SERVICE UNIT.	FAA AD 94-03-05 AND BOEING TELEX M7240-93-1411 DT. JULY 20 '1993	AS IN AD AND TELEX.	AS IN AD AND TLEX.
174	DGCA/BOEING 747/174	INSTALLATION OF AN ALUMINIUM SEAL RETAINER ON THE BULB SEAL TO THE FASCIA PANEL OF THE ESCAPE SYSTEMS ON SUD.	FAA AD 94-01-05 ; BO. ASB 747-25A3056	AS IN AD AND ASB.	AS IN AD AND ASB.
175	DGCA/BOEING 747/175	VISUAL INSPECTION/REPLACEMENT OF H-11 BOLTS.	FAA AD 94-07-06 AND BOEING S.B 747-51-2048 R1 DT. JAN 27 '1994.	AS IN AD AND SB.	AS IN AD AND SB.
176	DGCA/BOEING 747/176	REPLACEMENT - CERTAIN PNEUMATIC DUCT COUPLINGS WITH REDESIGNED COUPLINGS.	FAA AD 94-09-13 AND PUROLATOR SB FSC-912	AS IN AD AND SB.	AS IN AD AND SB.

177	DGCA/BOEING 747/177	INSPECTION - FORWARD LOWER ENGINE MOUNT BOLTS FOR MIGRATION AND DAMAGE.	FAA AF 94-10-05 AND BOEING SB 747-71-2192	AS IN AD AND SB.	AS IN AD AND SB.
178	DGCA/BOEING 747/178	MODIFICATION OF THE THRUST REVERSER CONTROL SYSTEM BY INSTALLING SOLENOID OPERATED SHUT OFF VALVE.	FAA AD 94-10-10 AND BOEING SB 747-78-2052 LATEST REV.	AS IN AD AND SB.	AS IN AD AND SB.
179	DGCA/BOEING 747/179R1	TO PREVENT FATIGUE CRACKING IN CERTAIN LAP JOINTS WHICH COULD RESULT IN RAPID DEPRESSURIZATION OF THE AIRPLANE	FAA AD 2010-10-05	AS IN AD AND SB.	AS IN AD AND SB.
180	DGCA/BOEING 747/180	TO PREVENT RAPID DECOMPRESSION OF THE AIRPLANE AND THE INABILITY TO CARRY FAIL SAFE LOADS.	FAA AD 94-12-09 AND BOEING SB 747-53-2307 R2	AS IN AD AND SB.	AS IN AD AND SB.
181	DGCA/BOEING 747/181	ATA 53 - INSPECTION FOR CRACKS IN FUSELAGE UPPER DECK TENSION TIES AND REPAIR OR MODIFICATION.	FAA AD 94-13-06 AND BOEING SB 747-53-2371	AS IN AD AND SB.	AS IN AD AND SB.
182	DGCA/BOEING 747/182	ATA 53 - HIGH FREQUENCY EDDY CURRENT INSPECTION FOR CRACKS AT UPPER ROW OF COUNTERSUNK FASTENERS IN LONGITUDINAL LAP JOINTS.	FAA AD 94-15-17 AND BOEING SB 747-53-2366	AS IN AD AND SB.	AS IN AD AND SB.
183	DGCA/BOEING 747/183R1	TO DETECT & CORRECT FATIGUE CRACKING IN THE FUSELAGE LOWER LOBE LONGITUDINAL LAP JOINTS WHICH COULD LEAD TO RAPID DECOMPRESSION OF THE AIRPLANE AND THE INABILITY OF THE STRUCTURE TO CARRY FAIL SAFE LOADS	FAA AD 2010-14-10	AS IN AD.	AS IN AD.
184	DGCA/BOEING 747/184	TO PREVENT SEPERATION OF THE STRUT FROM THE WING OF THE AIRPLANE DUE TO FATIGUE CRACKING.	FAA AD 94-17-17	AS IN AD.	AS IN AD.
185	DGCA/BOEING 747/185	TO PREVENT DELAYED INFLATION OF EVACUATION RAMP/SLIDES.	FAA AD 94-17-09 SB 7A/418-25-253 OR SB 7A/410-25-253 REV2.	AS IN AD & SB.	AS IN AD & SB.
186	DGCA/BOEING 747/186	TO PREVENT SEPERATION OF AN ENGINE DUE TO OVERHEATING & SUBSEQUENT CRACKING OF ENGINE STRUT.	FAA AD 94-22-08 & SB 747-54-2091.	AS IM AD & SB.	AS IN AD & SB.
187	DGCA/BOEING 747/187	TO PREVENT FAILURE OF A SPRING BEAM.	FAA AD 98-21-17 & BOEING ASB 747-54A 2171 DT. 31 OCT 94.	SA IN AD AND ASB.	AS IN AD AND ASB.
188	DGCA/BOEING 747/188	INSTALLATION OF A SEAL ON THE WING FRONT SPAR AT ENGINE STRUT.	FAA AD 98-21-17	AS IN AD& SB.	AS IN AD & SB.
189	DGCA/BOEING 747/189	TO ENSURE PROPER DISTRIBUTION OF THE FIRE EXTINGUISHER AGENT WITH IN THE NACELLE IN THE EVENT OF A FIRE.	FAA AD 95-02-13 & BOEING ASB 747-26-A2226 DT. JUNE 30, 1994 OR REV1 DT. NOV 28, 1994 .	AS IN AD & SB.	AS IN AD & SB.
190	DGCA/BOEING 747/190	WINDSHEAR DETECTION RECOVERY GUIDANCE SYSTEM (WSS).	FAA AD 95-04-01.	AS IN AD.	AS IN AD.
191	DGCA/BOEING 747/191	TO PREVENT LOSS OF AIR PRESSURE.	FAA AD 95-04-08	AS IN AD.	AS IN AD.
192	DGCA/BOEING 747/192	TO PREVENT FAILURE OF STRUT & SUBSEQUENT LOSS OF ENGINE.	FAA AD 98-08-15	AS IN AD & SB.	AS IN AD & SB.
193	DGCA/BOEING 737/193	TO PREVENT EXPLOSIVE FAILURE OF THE POTABLE WATER TANK.	FAA AD 95-11-03.	AS IN AD.	AS IN AD.

194	DGCA/BOEING 747/194	TO PREVENT FAILURE OF STRUT & SUBSEQUENT LOSS OF ENGINE.	FAA AD 95-13-06 & SB 74754A2156 DT.15 DEC 1994.	AS IN AD.	AS IN AD.
195	DGCA/BOEING 747/195	TO PREVENT STRUCTURAL FAILURE OF LONGITUDNAL FLOOR BEAM & KEEL BEAM & THE SUBSEQUENT RUPTURE OF FUSELAGE.	T95-15-52.	AS IN TAD.	AS IN TAD.
196	DGCA/BOEING 747/196	TO ENSURE INTEGRITY OF THE FAILSAFE FEATURES OF THE THRUST REVERSER SYSTEM.	FAA AD 2003-20-16	AS IN AD & ASB.	AS IN AD & ASB
197	DGCA/BOEING 747/197	TO PREVENT STRUCTURAL COLLAPSE AND SEPERATION OF THE AFT FUSELAGE FROM THE AIRPLANE.	FAA AD 96-01-03.	AS IN AD.	AS IN AD.
198	DGCA/BOEING 747/198	TO PREVENT MALFUNCTION OF THE SAFETY INTERLOCK SYSTEM OF THE MAIN DECK CARGO DOOR AND SUBSEQUENT RAPID DECOMPRESSION OF AIRCRAFT.	FAA AD 96-01-03.	AS IN AD.	AS IN AD
199	DGCA/BOEING 747/199	TO PREVENT SEPERATION OF THE ENGINE FROM THE AIRPLANE.	FAA AD 99-20-09	AS IN AD.	AS IN AD.
200	DGCA/BOEING 747/200	TO ENSURE THE INTEGRITY OF THE FAIL SAFE FEATURES OF THE THRUST REVERSER SYSTEM.	FAA AD 2000-15-04	AS IN AD.	AS IN AD.
201	DGCA/BOEING 747/201	TO ENSURE FLIGHT CREW AWARENESS OF THE CONFIGURATION OF THE AUTOLAND SYSTEM IN THE EVENT OF CHANGE FROM FAIL OPERATIONAL TO FAIL PASSIVE MODE.	FAA AD 96-06-06.	AS IN AD.	AS IN AD.
202	DGCA/BOEING 747/202	TO ENSURE THAT HINGE BOLTS ARE INSTALLED IN THE OVERHEAD STOWAGE PINS.	FAA AD 96-07-10.	AS IN AD.	AS IN AD.
203	DGCA/BOEING 747/203	TO PREVENT LEAKAGE OF OXYGEN FROM THE PASSENGER OXYGEN SUPPLY LINES.	FAA AD 96-14-01.	AS IN AD.	AS IN AD.
204	DGCA/BOEING 747/204	TO PREVENT FAILURE OF THE HYDRAULIC FUSE WHICH COULD RESULT IN THE FAILURE OF ONE OR MORE HYDRAULIC SYSTEMS & RESULTANT REDUCED CONTROLLABILITY OF THE AIRPLANE.	FAA AD 96-22-11.	AS IN AD.	AS IN AD.
205	DGCA/BOEING 747/205	TO PREVENT UNCOMMANDED ELEVATOR DEFLECTION, WHICH COULD RESULT IN STRUCTURAL DAMAGE AND REDUCED CONTROLLABILITY OF THE AIRPLANE.	FAA AD 96-22-01.	AS IN AD.	AS IN AD.
206	DGCA/BOEING74 7/206(R1)	TO PREVENT RAPID DECOMPRESSION OF THE AIRPLANE DUE TO DISBONDING AND SUBSEQUENT CRACKING OF THE SKIN PANELS.	FAA AD 2006-20-02 (FAA AD SUPERSEDES FAA AD 96-23-02)	AS IN AD.	AS IN AD.
207	DGCA/BOEING 747/207	TO DETECT AND CORRECT CORROSION ON GIRT BAR SUPPORT FITTINGS, WHICH COULD RESULT IN SEPARATION OF THE ESCAPE SLIDES AT THE MAIN ENTRY DOORS DURING AN EMERGENCY.	FAA AD 2007-12-11 SUPERCEDED FAA AD 96-23-05.	AS IN AD.	AS IN AD.
208	DGCA/BOEING 747/208	TO PREVENT FAILURE OF THE ENGINE PYLONAND CONSEQUENT SEPERATION OF THE ENGINE FROM THE WING DUE TO MIGRATION OF THE FUSE PINS INSTALLED AT THE MIDSPAR/SPRING BEAM FITTINGS AT THE PYLON.	FAA AD 96-26-52R1	AS IN AD.	AS IN AD.
209	DGCA/BOEING 747/209	TO PREVENT DROOPING OF THE STRUT AT THE STRUT -TOOWING INTERFACE AND CONSEQUENT SEPARATION OF THE ENGINE AND STRUT FROM THE AIRPLANE DUE TO CRACKING OR FRACTURING OF THE MIDSPAR FITTING CLEVIS.	FAA AD 96-25-01	AS IN AD.	AS IN AD.
210	DGCA/BOEING 747/210	TO DETECT AND CORRECT FATIGUE CRACKING OF NWW VERTICAL BEAMS AND FRAMES	FAA AD 2001-14-18	AS IN AD.	AS IN AD.

211	DGCA/BOEING 747/211	TO DETECT AND CORRECT ABRASION OF THE TEFLON SLEEVING AND WIRES ON THE BUNDLES OF FUEL BOOST PUMPS FOR THE NUMBERS 1 AND 4 MAIN FUEL TANKS.	FAA AD 97-26-07	AS IN AD.	AS IN AD.
212	DGCA/BOEING 747/212	TO PREVENT FUEL LEAKAGE AT THE FUEL BOOST AND OVERRIDE/JETTISON PUMPS.	FAA AD 97-03-17	AS IN AD.	AS IN AD.
213	DGCA/BOEING 747/213	TO PREVENT LEAKAGE OF FUEL INTO FORWARD CARGO BAY	FAA AD 2000-11-07	AS IN AD	AS IN AD
214	DGCA\BOEING 747/214	REPLACEMENT OF CERTAIN SWITCHES IN THE CABIN ATTENDANT'S PANEL	FAA AD 97-08-05 & BOEING ASB 747-33A2252 DT.1.8.96	AS IN AD & ASB	AS IN AD & ASB
215	DGCA\BOEING 747/215	TO DETECT & CORRECT CORROSION & CONSEQUENT FATIGUE CRACKING OF UPPER DECK FLOW BEAM AT STATION.	FAA AD 97-09-13	AS IN AD	AS IN AD
216	DGCA\BOEING 747/216	TO PREVENT UNCOMMANDED MOVEMENT OF THE PILOTS SEATS DURING ACCELERATION & TAKE OFF OF AIR PLANE	FAA AD 98-03-10.	AS IN AD	AS IN AD
217	DGCA\BOEING 747/217	TO DETECT & CORRECT FATIGUE CRACKING IN THE LOWER SPAN FITTING PLUG ON THE LOWER SPAN FITTING BODY, WHICH COULD RESULT IN FAILURE OF SHUT & SEPARATION OF ENGINE FROM THE AIRPLANE.	FAA AD 97-20-01	AS IN AD	AS IN AD
218	DGCA\BOEING 747/218	TO PREVENT POTENTIAL FAILURE WITHIN THE ELECTRICAL MOTOR ASSEMBLY OF THE SCVENGE PUMP.	FAA AD 98-14-17	AS IN AD.	AS IN AD.
219	DGCA/BOEING 747/219	TO DETECT AND CORRECT FATIGUE CRACKING OF THE LONGERON SPLICE FITTINGS AT STRINGER 11, WHICH COULD RESULT IN REDUCED CONTROLLABILITY AT THE HORIZONTAL STABILIZER.	FAA AD 2000-10-23	AS IN AD	AS IN AD
220	DGCA/BOEING 747/220	TO PREVENT ENGINE FLAME OUT AND CONSEQUENT SHUTDOWN DUE TO THE USE OF JP4 OR 2B FUEL.	FAA AD 98-08-23	AS IN AD.	AS IN AD.
221	DGCA/BOEING 747/221	TO DETECT AND CORRECT FATIGUE CRACKING IN THE INTERNAL SYSTEM.	FAA AD 98-07-15	AS IN AD.	AS IN AD.
222	DGCA/BOEING 747/222(R1)	INSPECTION TO DETECT FATIGUE CRACKING IN UPPER DECK FLOOR BEAMS.	FAA AD 2005-07-21 (ORIGINAL AD 98-09-17 IS SUPERSEDED) (SB 747-53A2420)	AS IN AD.	AS IN AD.
223	DGCA/BOEING 747/223	TO PREVENT LOSS OF RUDDER CONTROL	FAA AD 98-13-12	AS IN AD.	AS IN AD.
224	DGCA/BOEING 747/224	TO PREVENT ACCELERATED FATIGUE CRACKING ON INNER CHOC OF THE BS 2598 BULKHEAD.	FAA AD 98-16-13	AS IN AD	AS IN AD.
225	DGCA/BOEING 747/225	HAZARDS OF DRY OPERATION TO OVERRIDE JETTISON PUMP OF THE CENTRE WING FUEL TANK.	FAA AD 98-16-19	AS IN AD.	AS IN AD.
226	DGCA/BOEING 747/226	TO PREVENT HIGH FUEL PRESSURE IN COMPONENTS BETWEEN THE FUEL SHUT OFF SPAR VALVE & THE ENGINE FUEL SHUTOFF VALVE.	FAA AD 2000-14-06	AS IN AD	AS IN AD
227	DGCA/BOEING 747/227	TO PREVENT INSTALLATION AND USE OF SWITCHES IN THE CABIN ATTENDED.	FAA AD 98-20-37.	AS IN AD.	AS IN AD.

228	DGCA/BOEING 747/228	TO PREVENT ELECTRICAL TRANSIENT INDUCED BY EMI.	FAA AD 98-20-40.	AS IN AD.	AS IN AD.
229	DGCA/BOEING 747/229	TO PREVENT CONTACT BETWEEN THE ROTATING PADDLE WHEELAND THE STATIONARY END PLATES.	FAA AD 2001-13-12	AS IN AD	AS IN AD
230	DGCA/BOEING 747/230	REPETITIVE INSPECTION OF THE PUTBOARD NACELLE STRUTS	FAA AD 99-04-01	AS IN AD	AS IN AD
231	DGCA/BOEING 747/231	TO PREVENT FATIGUE CRACKING & LOOSE OR MISSING FASTENERS IN THE AFT TORQUE BULKHEADS OF THE OUTBOARD NACELLE STRUTS	FAA AD 2000-12-16	AS IN AD	AS IN AD
232	DGCA/BOEING 747/232	ONE TIME DETAILED VISUAL INSPECTION OF THE OUTBOARD SEQUENCE CARRIAGE ATTACHMENT FITTING	FAA AD 99-05-02	AS IN AD	AS IN AD
233	DGCA/BOEING 747/233	REPETITIVE INSPECTION OF A NEW AREA OF THE DIAGONAL BRACE OF THE ENGINE STRUT	FAA AD 99-01-10	AS IN AD	AS IN AD
234	DGCA/BOEING74 7/234(R1)	TO DETECT AND CORRECT STRESS CORROSION OF THE STOP SUPPORT FITTINGS OF THE MAIN ENTRY DOORS	FAA AD 2005-10-18 (FAA AD 98-26-13 IS SUPERSEDED)	AS IN AD	AS IN AD
235	DGCA/BOEING 747/235	REPETITIVE DETAILED VISUAL INSPECTION AT BODY STATION 1241	FAA AD 98-26-23	AS IN AD	AS IN AD
236	DGCA/BOEING 747/236	REPETITIVE VISUAL INSPECTIONS FOR CRACKS & HIGH FREQUENCY EDDY CURRENT (HFEC)INSPECTIONS ON NO.2 &3 ENGINE STRUTS	FAA AD 99-08-01	AS IN AD	AS IN AD
237	DGCA/BOEING 747/237	INSPECTIONS TO DETECT DISCREPANCIES OF THE CENTER FUEL TANK WIRING & COMPONENTS & ELECTRICAL BONDING TEST OF THE CENTER FUEL TANK COMPONENTS	FAA AD 99-08-02 R1	AS IN AD	AS IN AD
238	DGCA/BOEING 747/238	INSPECTIONS TO DETECT CRACKS IN THE EDGE FRAME WEB & DOUBLER OF THE NO.1 MAIN ENTRY DOOR CUT OUT	FAA AD 99-07-03	AS IN AD	AS IN AD
239	DGCA/BOEING 747/239	INSPECTIONS TO DETECT CRACKING OR FRACTURE OF THE STEEL ATTACHMENT FITTINGS OF THE DIAGONAL BRACE TO THE NACELLE STRUTS	FAA AD 2001-16-08	AS IN AD	AS IN AD
240	DGCA/BOEING 747/240	TO PREVENT AIRFRAME DAMAGE AS A RESULT OF BLUE ICE	FAA AD 99-08-10	AS IN AD	AS IN AD
241	DGCA/BOEING 747/241	TO DETECT & CORRECT FATIGUE CRACKING OF THE WING FRONT SPAR WEB	FAA AD 99-10-09	AS IN AD	AS IN AD
242	DGCA/BOEING 747/242	TO DETECT & CORRECT AN IMPROPERLY INSTALLED AILERON CABLE,WHICH COULD LEAD TO THE FAILURE OF THE AILERON CABLE & CONSEQUENT REDUCED LATERAL CONTROL CAPABILITY OF THE AIRPLANE	FAA AD 99-11-15	AS IN AD	AS IN AD
243	DGCA/BOEING 747/243	TO PREVENT CRACKING OF CORROSION OF THE FUSE PIN OF THE NACELLE STRUT	FAA AD 99-10-10	AS IN AD	AS IN AD
244	DGCA/BOEING 747/244	TO DETECT & REPAIR CRACKING OF THE E-42 SATCOM RACK & ITS SUPPORTING STRUCTURE	FAA AD 99-14-04	AS IN AD	AS IN AD
245	DGCA/BOEING 747/245	TO PREVENT DETERIORATION OF CABLE GUARDS IN THE LATERAL CONTROL SYSTEM	FAA AD 99-24-16	AS IN AD	AS IN AD

246	DGCA/BOEING 747/246	TO DETECT & CORRECT FATIGUE CRACKING OF THE UPPER & LOWER CHORDS OF THE WING FRONT SPAR	FAA AD 2000-03-01	AS IN AD	AS IN AD
247	DGCA/BOEING 747/247	TO PREVENT IMPROPER LATCHING OF LATCH PINS & THE MATING LATCH CAM ON THE CARGO DOOR	FAA AD 2000-02-37	AS IN AD	AS IN AD
248	DGCA/BOEING 747/248	TO PREVENT LOSS OF THE UNDERWING FITTING LOAD PATH DUE TO MISING DAMAGED OR BROKEN TAPER LOCK BOLTS	FAA AD 2000-03-22	AS IN AD	AS IN AD
249	DGCA/BOEING74 7/249(R1)	TO PREVENT FAILURE OF THE UPPER DECK FLOOR BEAMS DUE TO FATIGUE CRACKING	FAA AD 2005-06-11 (ORIGINAL AD 2000-04-17 IS SUPERSEDED)	AS IN AD	AS IN AD
250	DGCA/BOEING 747/250	TO PREVENT ENGINE THRUST CONTROL CABLE FAILURES	FAA AD 2000-05-30	AS IN AD	AS IN AD
251	DGCA/BOEING 747/251	TO DETECT & CORRECT CRACKING OF THE INNER CHORD & WEB OF THE BODY STATION 1265 EDGE FRAME BETWEEN STRINGERS 23 & 27	FAA AD 2000-02-10	AS IN AD	AS IN AD
252	DGCA/BOEING 747/252	TO PREVENT CRACKS IN THE FIRE EXTINGUISHING DISCHARGE TUBES IN THE ENGINE STRUTS.	FAA AD 2002-19-12	AS IN AD	AS IN AD
253	DGCA/BOEING74 7/253(R2)	TO PREVENT FATIGUE CRACKING OF THE BODY STATION (BS) 2598 BULKHEAD STRUCTURE.	FAA AD 2010-14-07	AS IN AD	AS IN AD
254	DGCA/BOEING 747/254	REPETITIVE INSPECTIONS OR CHECKS TO DETECT BROKEN H-11 STEEL BOLTS	FAA AD 2000-08-14	AS IN AD	AS IN AD
255	DGCA/BOEING 747/255	TO PREVENT SEPARATION OF THE NOSE COWL FROM THE ENGINE	FAA AD 2000-09-08	AS IN AD	AS IN AD
256	DGCA/BOEING 747/256	TO DETECT & CORRECT CRACKING IN THE AFT PRESSURE BULK HEAD	FAA AD 2000-12-19	AS IN AD	AS IN AD
257	DGCA/BOEING 747/257	TO DETECT & CORRECT FATIGUE CRACKING OF CERTAIN AREAS OF THE FUSELAGE SKIN	FAA AD 2000-14-04	AS IN AD	AS IN AD
258	DGCA/BOEING 747/258	TO PREVENT HIGH VELOCITY SEPARATION OF A BRAKE SYSTEM ACCUMULATOR BARREL,PISTON OR END CAP	FAA AD 2000-14-01	AS IN AD	AS IN AD
259	DGCA/BOEING 747/259	TO DETECT & CORRECT BUSHING MIGRATION,CORROSION OR CRACKING OF THE LUGS ON THE BULKHEAD FITTING OF THE REAR ENGINE MOUNT	FAA AD 2001-15-15	AS IN AD	AS IN AD
260	DGCA/BOEING 747/260	TO ENSURE THE INTEGRITY OF THE FAIL SAFE FEATURES OF THE THRUST REVERSER SYSTEM	FAA AD 2000-17-06	AS IN AD	AS IN AD
261	DGCA/BOEING 747/261	TO PREVENT CONTACT BETWEEN THE PRECOOLER SUPPORT LINK & THE PRECOOLER SUPPORT FITTING	FAA AD 2000-21-06	AS IN AD	AS IN AD
262	DGCA/BOEING 747/262	TO PREVENT CHAFING & FAILURE OF ENGINE THRUST CONTROL CABLES	FAA AD 2000-22-15	AS IN AD	AS IN AD
263	DGCA/BOEING 747/263	TO PREVENT LOOSE FASTENERS AND ASSOCIATED DAMAGE WHICH COULD RESULT IN SEPARATION OF ENGINE	FAA AD 2007-19-19 SUPERSEDES FAA AD 2001-15-02	AS IN AD	AS IN AD

264	DGCA/BOEING 747/264	TO DETECT & CORRECT CRACKING OF THE VERTICAL CHORDS ADJACENT TO THE LOWER SPAR FITTING	FAA AD 2001-12-28	AS IN AD	AS IN AD
265	DGCA/BOEING 747/265	TO PREVENT FATIGUE CRACKING OF THE FRAME WEB & DOUBLER OF THE FORWARD EDGE FRAME OF MAIN ENTRY DOOR NO.1	FAA AD 2000-24-07	AS IN AD	AS IN AD
266	DGCA/BOEING 747/266	TO DETECT & CORRECT FATIGUE CRACKING OF THE FRONT SPAR WEB OF THE WING	FAA AD 2003-16-12	AS IN AD	AS IN AD
267	DGCA/BOEING 747/267	TO PREVENT POTENTIAL IGNITION OF FIBERGLASS INSULATION IN THE ENVIRONMENTAL CONTROL SYSTEM (ECS) DUCTS	FAA AD 2000-26-05	AS IN AD	AS IN AD
268	DGCA/BOEING74 7/268(R1)	TO PREVENT POTENTIAL IGNITION OF THE MOISTURE BARRIER COVER OF THE DRIP SHIELD	FAA AD 2005-17-09 (FAA AD 2000-26-04 IS SUPERSEDED)	AS IN AD	AS IN AD
269	DGCA/BOEING 747/269	TO PREVENT LOSS OF A FLAP TRANSMISSION	FAA AD 2001-03-10	AS IN AD	AS IN AD
270	DGCA/BOEING 747/270	TO FIND & FIX DISCREPANCIES OF THE INSTALLATION OF THE MID SPAR FUSE PINS OF THE INBOARD & OUTBOARD STRUT	FAA AD 2001-23-15	AS IN AD	AS IN AD
271	DGCA/ BOEING 747/271	CHEMICAL OXYGEN GENERATOR	FAA AD 2001-10-14	AS IN AD	AS IN AD
272	DGCA/BOEING 747/272	PREVENTION OF UNRESTRAINED MOVEMENT OF PASSENGER SEATS DURING HIGH FWD DECELERATION OF AIRPLANE	FAA AD 2001-11-11	AS IN AD	AS IN AD
273	DGCA/BOEING 747/273	TODETECT AND CORRECT HEAT DAMAGE TO THE DIAGONAL BRACE	FAA AD 2001-26-12	AS IN AD	AS IN AD
274	DGCA/BOEING 747/274	TO DETECT AND CORRECT FATIGUE CRACKING OF THE SKIN	FAA AD 2001-11-06	AS IN AD	AS IN AD
275	DGCA/BOEING 747/275	CANCELLED IN VIEW OF DGCA/B 747/253 (R1) VIDE LETTER NO 9-456/06 AI(2) DATED 27-03-06			
276	DGCA/BOEING 747/276	TO REMOVE POWER FROM THE IFE SYSTEM WHEN NECESSARY	FAA AD 2001-14-11	AS IN AD	AS IN AD
277	DGCA/BOEING74 7/277	TO DECECT AND CORRECT FATIGUE CRACKS THAT COULD EXTEND AND FULLY SEVERE THE STATION 800 FRAME ASSEMBLY	FAA AD 2007-16-08 SUPERSEDED FAA AD 2006-12-12 (FAA AD 2001-14-22 IS SUPERSEDED)	AS IN AD	AS IN AD
278	DGCA/BOEING74 7/278(R2)	AIR TRANSPORT ASSOCIATION (ATA) OF AMERICA CODE 53: FUSELAGE	FAA AD 2010-20-08	AS IN AD	AS IN AD
279	DGCA/BOEING 747/279(R1)	TO PREVENT CRACKING AND OTHER DAMAGE OF THE ACTUATOR ATTACH FITTING OF THE TRAILING EDGE FLAP	FAA AD 2005-20-18 (FAA AD 2001-13-12 IS SUPERSEDED)	AS IN AD	AS IN AD
280	DGCA/BOEING74 7/280(R1)	TO PREVENT LOSS OF THE UNDERWING FITTING LOAD PATH DUE TO MISSING OR DAMAGED ALLOY STEEL OR A286 TAPERLOCK FASTERNES.	FAA AD 2005-14-08 (FAA AD 2001-13-06 IS SUPERSEDED)	AS IN AD	AS IN AD
281	DGCA/BOEING 747/281	TO PREVENT INADVERTOUL DEPLOYMENT OF THRUST REVERSER	FAA AD 2000-01-05 & FAA AD 2000-10-17	AS IN AD	AS IN AD

282	DGCA/BOEING 747/282	TO PREVENT POTENTIAL FAILURE WITHIN THE ELECTRICAL MOTOR ASSEMBLY	FAA AD 99-08-19	AS IN AD	AS IN AD
283	DGCA/BOEING74 7/283(R2)	TO DETECT AND CORRECT FATIGUE CRACKING OF CERTAIN LOWER LOBE FUSELAGE FRAMES	FAA AD 2006-09-06 R1 (FAA AD 99-07-12 IS SUPERSEDED)	AS IN AD	AS IN AD
284	DGCA/BOEING 747/284	TO DETECT AND CORRECT FATIGUE CRACKING OF FUEL ENGINE	FAA AD 98-02-02	AS IN AD	AS IN AD
285	DGCA/BOEING 747/285	TO PREVENT LOSS OF RUDDER CONTROL DUE TO INPROPERLY	FAA AD 2001-22-13	AS IN AD	AS IN AD
286	DGCA/BOEING 747/286	TO PREVENT DAMAGE TO THE FLAP SYSTEM, ADJACENT SYSTEMS OR STRUCTURAL COMPONENTS	FAA AD 2001-23-13	AS IN AD	AS IN AD
287	DGCA/BOEING74 7/287	TO PREVENT A SHOPRT CIRCUIT AND RESULTANT ARCING	FAA AD 2001-24-33	AS IN AD	AS IN AD
288	DGCA/BOEING 747/288	TO PREVENT CHAFINING BETWEEN OXYGEN HOSE AND ELECTRICAL WIRE BUNDLES	FAA AD 2001-24-29	AS IN AD	AS IN AD
289	DGCA/BOEING 747/289	TO PREVENTCHAFINING AND BURNING OF ELECTRICAL WIRES	FAA AD 2001-24-31	AS IN AD	AS IN AD
290	DGCA/BOEING 747/290	TO PREVENT CRACKING OF THE VERTI CAL CHORDS ADJCENT RO THE LOWER SPAR FITTING	FAA AD 2002-05-01	AS IN AD	AS IN AD
291	DGCA/BOEING74 7/291(R1)	TO DETECT AND CORRECT CRACKING IN THE OUTBOARD AND CENTER SECTION OF THE HORIZONTAL STABILIZER	FAA AD 2006-10-16 (FAA AD'S 2002-06-02 AND 2003-13-09 ARE SUPERSEDED)	AS IN AD	AS IN AD
292	DGCA/BOEING74 7/292	TO PREVENT CRACKED OR BROKEN BOLTS WHICH COULD RESULT IN STRUCTURAL DAMAGE AND RAPID DEPRESSURIZATION	FAA AD 2002-26-17	AS IN AD	AS IN AD
293	DGCA/BOEING74 7/293	TO PREVENT FUEL VAPOURS FROM COMING INTO CONTACT WITH AN IGNITION SOURCE IN THE CENTER WING FUEL TANK	EMERGENCY FAA ADs 2002-19-52 & 2002-24-51	AS IN AD	AS IN AD
294	DGCA/BOEING/74 7/294	TO FIND AND CRACKING OF THE LOWER LOBE PANEL OF THE FUSELAGE SKIN OF THE AFT CARGO BAY	FAA AD 2002-23-15	AS IN AD	AS IN AD
295	DGCA/BOEING/74 7/295	TO FIND AND FIX DISCREPANT FASTENERS OF THE DIAGONAL BRACE FITTINGS	FAA AD 2002-24-03	AS IN AD	AS IN AD
296	DGCA/BOEING74 7/296	TO DETECT AND CORRECT WEAR DAMAGE OF THE FUSELAGE SKIN IN THE INTERFACE AREA OF THE VERTICAL STABILIZER SEAL & FUSELAGE SKIN IN SECTIONS 46 & 48.	FAA AD 2009-14-02	AS IN AD	AS IN AD
297	DGCA/BOEING74 7/297	TO PREVENT INCAPACIATION OF THIS FLIGHT CREW DUE TO LACK OF OXYGEN	FAA AD 2003-03-15R1	AS IN AD	AS IN AD
298	DGCA/BOEING74 7/298	TO FIND AND FIX POSSIBLE FATIGUE CRACKING OF THE FUSELAGE SKIN	FAA AD 2003-03-19	AS IN AD	AS IN AD
299	DGCA/BOEING74 7/299	TO PREVENT ABNORMAL OPERATION OR RETRACTION OF A TRAILING EDGE FLAP	FAA AD 2003-08-11	AS IN AD	AS IN AD

300	DGCA/BOEING747/300	TO FIND AND FIX CRACKING OF THE SKIN, BEAR STRAP, AND SILL CHORS OF THE LOWER CARGO DOOR CUTOUT	FAA AD 2003-10-06	AS IN AD	AS IN AD
301	DGCA/BOEING747/301	TO PREVENT BROKEN H-11 STEEL BOLTS	FAA AD 2003-11-25	AS IN AD	AS IN AD
302	DGCA/BOEING747/302	CANCELLED IN VIEW OF DGCA/B 747/291 R1 VIDE LETTER NO 9-456/06 AI(2) DATED 25-05-06			
303	DGCA/BOEING747/303	TO PREVENT THE LOSS OF AN INBOARD TRAILING EDGE FOREFLAP DURING FLIGHT	FAA AD 2003-16-08	AS IN AD	AS IN AD
304	DGCA/BOEING747/304	TO PREVENT FAILURE OF THE STRUT AND SUBSEQUENT LOSS OF THE ENGINE	FAA AD 95-10-16	AS IN AD	AS IN AD
305	DGCA/BOEING747/305	TO PREVENT FAILURE OF THE STRUT AND SEPARATION OF AN ENGINE FROM THE AIRPLANE DUE TO FRACTURING OF THE FUSE PINS.	FAA AD 97-14-06	AS IN AD	AS IN AD
306	DGCA/BOEING747/306	INSPECTION OF UPPER AND LOWER SURFACE OF THE FORWARD LOWER SPAR	FAA AD 2003-16-14	AS IN AD	AS IN AD
307	DGCA/BOEING747/307	MODIFICATION OF THE ESCAPE SLIDE / RAFT PACK ASSEMBLY AND CABLE RELEASE SLIDERS	A.S.B.747-25-3274R1 & A.S.B.747-25-3307	AS IN SB AND WITHIN 3 MNTHS OR AVAILIBILITY OF SPARES (which ever is earlier)	AS IN SB
308	DGCA/BOEING747/308	CANCELLED IN VIEW OF DGCA/B 747/253 (R1) VIDE LETTER NO 9-456/06 AI(2) DATED 27-03-06			
309	DGCA/BOEING747/309(R1)	TO PREVENT DETACHMENT OF THE SHOULDER RESTRAINT HARNESS OF THE ATTENDANT OR OBSERVER SEAT	FAA AD 2006-26-13	AS IN AD & SB	AS IN AD & SB
310	DGCA/BOEING747/310	TO PREVENT LOSS OF A CARGO DOOR AND CONSEQUENT RAPID DEPRESSURIZATION OF THE AIRPLANE.	FAA AD 2003-25-11	AS IN AD & SB	AS IN AD & SB
311	DGCA/BOEING747/311	TO PREVENT LEAKAGE OF HYDRAULIC (FLAMMABLE) FLUID INTO AN ENGINE FIRE	FAA AD 2003-26-13	AS IN AD & SB	AS IN AD & SB
312	DGCA/BOEING747/312	TO FIND AND FIX FATIGUE CRACKING OF STRUCTURE NEAR AND COMMON TO THE UPPER CHORD AND SPLICE FITTINGS OF THE REAR SPAR WING	FAA AD 2004- 03- 09	AS IN AD & SB	AS IN AD & SB
313	DGCA/BOEING747/313 (R1)	TO FIND AND FIX CRACKING IN CERTAIN UPPER DECK FLOOR BEAMS	FAA AD 2006-08-02 (FAA AD 2004-03-11 IS SUPERSEDED)	AS IN AD & SB	AS IN AD & SB
314	DGCA/BOEING747/314	TO PREVENT THE FAILURE OF AN EMERGENCY EVACUATION SLIDE OR SLIDE / RAFT TO FULLY INFLATE DURING AN EMERGENCY SITUATION	FAA AD 2004-03-17	AS IN AD & SB	AS IN AD & SB
315	DGCA/BOEING747/315	CANCELLEN IN VIEW OF DGCA/BOEING747 /323 VIDE LETTER NO 9-456/04 DATED 01-04-05			
316	DGCA/BOEING747/316	TO PREVENT INPROPER CONNECTION OF THE LATCH ,WHICH COULD RESULT IN SEPARATION OF A COWL PANEL FROM THE AIRPLANE	FAA AD 2004-07-19 & ASB 747-71-2301	AS IN AD	AS IN AD

317	DGCA/BOEING74 7/317	TO ENSURE THE CONTINUED STRUCTURAL INTEGRITY	FAA AD 2004-07-22R1 REVISES FAA AD 2004-07-22	AS IN AD	AS IN AD
318	DGCA/BOEING74 7/318	TO PREVENT OVERHEATING OF THE HEATER TAPE ON POTABLE WATER FILL AND DRAIN LINES	FAA AD 2007-19-16 SUPERSEDES FAA AD 2004-09-10	AS IN AD & SB	AS IN AD & SB
319	DGCA/BOEING74 7/319	TO ENSURE ADEQUATE ELECTRICAL BONDING BETWEEN THE PENETRATION FITTINGS OF THE HYDRAULIC HEAT EXCHANGER AND REAR SPARS OF THE FUEL TANKS	FAA AD 2004-10-06	AS IN AD	AS IN AD
320	DGCA/BOEING74 7/320(R1)	TO DETECT AND CORRECT CRACKS IN THE OVERLAPPED SKIN PANELS IN THE FUSELAGE SKIN LAP JOINTS	FAA AD 2004-13-02 & FAA AD 2006-22-09	AS IN AD	AS IN AD
321	DGCA/BOEING74 7/321	INSPECTION OF THE FIRE EXTINGUISHER BOTTLES IN THE ENGINE AND THE APU, AND REPLACEMENT OF THE FIRE EXTINGUISHER BOTTLE.	FAA AD 2004-14-13 & ASB 747-26A2272	AS IN AD & AS IN ASB	AS IN AD & AS IN ASB
322	DGCA/BOEING74 7/322(R1)	TO PREVENT FRACTURE OF THE OUTER CYLINDER OF THE WING LANDING GEAR	FAA AD 2004-16-05	AS IN AD & SB	AS IN AD & SB
323	DGCA/BOEING74 7/323	TO DETECT AND CORRECT THE PROPAGATION OF FATIGUE CRACKS IN THE VICINITY OF OIL CANS ON THE WEB OF THE PRESSURE BULKHEAD	FAA AD 2009 - 04 -06	AS IN AD & ASB	AS IN AD & ASB
324	DGCA/BOEING74 7/324	TO PREVENT MINERAL BUILD UP ON THE APU AND ENGINE FIRE SHUTOFF SWITCHES.	FAA AD 2004-20-16&ASB747-26A2274	AS IN AD & ASB	AS IN AD & ASB
325	DGCA/BOEING74 7/325(R2)	TO DETECT & CORRECT CRACKS & FRACTURES OF THE NACELLE STRUT FRONT SPAR CHORD ASSEMBLY. FRACTURE OF THE FRONT SPAR CHORD ASSEMBLY COULD LEAD TO LOSS OF THE STRUT UPPER LINK LOAD PATH & CONSEQUENT FRACTURE OF THE DIAGONAL BRACE	FAA AD 2010 -01-10	AS IN AD & ASB	AS IN AD & ASB
326	DGCA/BOEING74 7/326	TO PREVENT UNEXPECTED DISCONNECT OF THE AUTOPILOT DURING OPERATION IN RVSM AIRSPACE DUE TO CLOSE PASSAGE OF ANOTHER AIRPLANE.	FAA AD 2004-25-06	AS IN AD	AS IN AD
327	DGCA/BOEING74 7/327(R2)	TO PREVENT FATIGUE CRACKS IN THE TOP AND SIDE PANEL WEBS AND STIFFNESS OF THE NOSE WHEEL WELL (NWW).	FAA AD 2006-22-15 (ORIGINAL AD 2004-25-23, 2005-09-02 ARE SUPERSEDED)	AS IN AD	AS IN AD
328	DGCA/BOEING74 7/328	TO DETECT AND CORRECT CRACKING IN THE MIDSPAR ASSEMBLY	FAA AD 2005-01-03& ASB 747-54A2219	AS IN AD& ASB	AS IN AD & ASB
329	DGCA/BOEING74 7/329	TO ENSURE ADEQUATE ELECTRICAL BONDING BETWEEN THE HOUSING OF EACH FUEL PUMP AND AIRPLANE STRUCTURE OUTSIDE THE FUEL TANKS.	FAA AD 2005-01-07& ASB 747-28A2033	AS IN AD& ASB	AS IN AD& ASB
330	DGCA/BOEING74 7/330	TO ENSURE THE STRUCTURAL INTEGRITY OF THE STRUT-TO-WING LOAD PATH AND PREVENT SEPERATION OF THE STRUT AND ENGINE FROM THE AIRPLANE	FAA AD 2005-03-01	AS IN AD	AS IN AD
331	DGCA/BOEING74 7/331	TO PREVENT ARCING OR SPARKING AT THE INTERFACE BETWEEN THE BULKHEAD FITTINGS OF THE ENGINE FUEL FEED TUBE AND THE FRONT SPAR INSIDE THE FUEL TANK OF THE WINGS AND BETWEEN THE OVERWING FUEL FILL PORTS AND THE AIRPLANE STRUCTURE DURING A LIGHTNING STRIKE.	FAA AD 2005-04-01	AS IN AD	AS IN AD
332	DGCA/BOEING74 7/332	CANCELLED IN VIEW OF DGCA/BOEING 747/76(R1) VIDE LETTER NO 9-456/04 AI(2) DATED 28/04/05 (FAA AD 2005-08-01 SUPERSEDED TWO ADS 91-11-01, AND 2005-04-51			

333	DGCA/BOEING74 7/333	TO DETECT AND CORRECT FATIGUE CRACKING OF THE WEB BULKHEAD.	FAA AD 2005-05-07 (ASB 747-54A2220)	AS IN AD	AS IN AD
334	DGCA/BOEING74 7/334	TO PREVENT FATIGUE CRACKING OF THE FUSELAGE FRAME TO TENSION TIE JOINTS.	FAA AD 2005-05-08 (BOEING SPL ATTENTION SB 747-53-2483)	AS IN AD	AS IN AD
335	DGCA/BOEING74 7/335	ONE TIME INSPECTION TO FIND AND FIX CRACKING IN CERTAIN UPPER DECK FLOOR BEAMS.	FAA AD 2005-06-12(BOEING SB 747-53A2459 REV 1 DATED 11 MAR 04)	AS IN AD & SB	AS IN AD & SB
336	DGCA/BOEING74 7/336	TO FIND AND FIX FATIGUE CRACKING OF THE FUSELAGE FRAME	FAA AD 2005-08-09 (ASB 747-53A2505)	AS IN AD & ASB	AS IN AD & ASB
337	DGCA/BOEING74 7/337	TO ENSURE THAT THE GALLEY CART LIFT CAN BE SENT ONLY FROM THE DECK ON WHICH IT IS IN USE.	FAA AD 2005-10-19	AS IN AD & ASB	AS IN AD & ASB
338	DGCA/BOEING74 7/338	TO DETECT AND CORRECT CRACKING OF THE C-3 FRAME UPPER CLOSURE FITTINGS	FAA AD 2005-10-22	AS IN AD & ASB	AS IN AD & ASB
339	DGCA/BOEING74 7/339	TO PREVENT THE OVERHEATING OF THE FREQUENCY CONVERTER'S UNDERSIZWD OUTPUT WIRING	FAA AD 2005-12-10	AS IN AD & SB	AS IN AD & SB
340	DGCA/BOEING74 7/340	TO DETECT AND CORRECT CRACKS OR MATERIAL LOSS IN THE FUSELAGE SKIN, AND CONSEQUENT REDUCED STRUCTURAL INTEGRITY OF THE SKIN PANEL	FAA AD 2005-15-06	AS IN AD & SB	AS IN AD & SB
341	DGCA/BOEING74 7/341R1	TO DETECT AND CORRECT FATIGUE CRACKING IN THE SPECIFIED FUSELAGE STRINGERS WHICH IF LEFT UNDETECTED COULD RESULT IN FUSELAGE SKIN CRACKING	FAA AD 2010-01-02	AS IN AD & SB	AS IN AD & SB
342	DGCA/BOEING74 7/342	TO PREVENT THE STRUCTURAL SUPPORT FOR THE DOOR 5 CREW REST AND ZONE E STOWBINS FAILING.	FAA AD 2005-16-01	AS IN AD & SB	AS IN AD & SB
343	DGCA/BOEING74 7/343	TO PREVENT ACTUATION DELAYS IN THE INFLATION SYSTEMS OF THE ESCAPE SLIDE/RAFTS	FAA AD 2009 -10-12	AS IN AD & SB	AS IN AD & SB
344	DGCA/BOEING74 7/344	TO DETECT AND CORRECT CRACKING OF THE AFT LOWER SPAR WEB	FAA AD 2005-16-11	AS IN AD & SB	AS IN AD & SB
345	DGCA/BOEING74 7/345(R1)	TO PREVENT CRACKED OR FRACTURED THRUST LINKS	FAA AD 2006-09-01 (FAA AD 2005-19-06 IS SUPERSEDED)	AS IN AD & SB	AS IN AD & SB
346	DGCA/BOEING74 7/346	TO PREVENT THE LOSS OF A DUAL SIDE BRACE (DSB) OR UNDERWING MIDSPAR FITTING LOAD PATH	FAA AD 2005-19-09	AS IN AD & SB	AS IN AD & SB
347	DGCA/BOEING74 7/347	TO DETECT AND CORRECT CRACKS IN THE FRAME WEB OF MAIN ENTRY DOOR NO 1.	FAA AD 2005-22-06	AS IN AD & SB	AS IN AD & SB
348	DGCA/BOEING74 7/348	TO FIND AND FIX CRACKS IN THE TENSION TIES AT THE BODY STATION 820 FRAME CONNECTION	FAA AD 2006-01-07	AS IN AD & SB	AS IN AD & SB
349	DGCA/BOEING74 7/349R1	TO PREVENT LOSS OF THE STRUCTURAL INTEGRITY OF THE FUSELAGE WHICH COULD RESULT IN RAPID DEPRESSURIZATION OF THE AIRPLANE	FAA AD 2010-01-01	AS IN AD & SB	AS IN AD & SB
350	DGCA/BOEING 747/350	TO PREVENT IMPROPER DEPLOYMENT OF THE ESCAPE SLIDE/RAFT OR BLOCKAGE OF THE PASSENGER/CREW DOORS IN THE EVENT OF AN EMERGENCY EVACUATION	FAA AD 2006-03-15	AS IN AD & SB	AS IN AD & SB

351	DGCA/BOEING 747/351	TO DETECT AND CORRECT FATIGUE CRACKS IN THE OVERLAPPING (UPPER) SKIN, UPPER FASTENER ROW OF THE LAP JOINTS OF THE FUSELAGE SKIN IN SECTIONS 41, 42, AND 46.	FAA AD 2006-05-09	AS IN AD & SB	AS IN AD & SB
352	DGCA/BOEING 747/352	TO DETECT AND CORRECT CRACKS IN THE SKIN, BEARSTRAP, AND SMALL STEEL DOUBLER (IF INSTALLED) AT MAIN ENTRY DOORS NO 3.	FAA AD 2006-06-10	AS IN AD & SB	AS IN AD & SB
353	DGCA/BOEING 747/353	TO DETECT AND CORRECT CRACKING OF THE TENSION TIES, SHEAR WEBS, AND FRAMES OF THE UPPER DECK WHICH COULD RESULT IN RAPID DECOMPRESSION & REDUCE STRUCTURAL INTEGRITY	FAA AD 2007 -23 -18 (FAA AD 2006 - 06 - 11 IS SUPERSEDED)	AS IN AD & SB	AS IN AD & SB
354	DGCA/BOEING 747/354	TO PREVENT FAILURE OF FLIGHT DECK ELECTRONIC EQUIPMENT	FAA AD 2005-05-20	AS IN AD & SB	AS IN AD & SB
355	DGCA/BOEING 747/355	TO PREVENT AN UNDETECTED FAILURE OF THE PRIMARY LOAD PATH FOR THE BALLSCREW IN THE HORIZONTAL STABILIZER	FAA AD 2006-10-02	AS IN AD & SB	AS IN AD & SB
356	DGCA/BOEING 747/356	TO DETECT AND CORRECT CRACKING IN THE LONGERON EXTENSION FITTING	FAA AD 2007-26-17 SUPERCEDED FAA AD 2006-10-04	AS IN AD & SB	AS IN AD & SB
357	DGCA/BOEING 747/357	TO ENSURE THAT THE MIDPIVOT BOLTS AND MIDPIVOT BOLT ACCESS DOORS ARE INSTALLED IN THE CORRECT POSITION	FAA AD 2006-12-03	AS IN AD & SB	AS IN AD & SB
358	DGCA/BOEING 747/358	TO DETECT AND CORRECT CRACKING, BUCKLING, WRINKLING, OR HEAT DAMAGE OF THE SKIN AND INTERNAL STRUCTURE OF THE ENGINE STRUT	FAA AD 2006-16-10	AS IN AD & SB	AS IN AD & SB
359	DGCA/BOEING 747/359	TO PREVENT RAPID DECOMPRESSION OF THE AIRPLANE DUE TO DISBONDING AND SUBSEQUENT CRACKING OF THE SKIN PANELS	FAA AD 2006-24-02	AS IN AD & SB	AS IN AD & SB
360	DGCA/BOEING 747/360	TO DETECT AND CORRECT FATIGUE CRACKS AT THE FASTENER ROWS OF THE FUSELAGE SKIN IN SECTION 41	FAA AD 2006-24-05	AS IN AD & SB	AS IN AD & SB
361	DGCA/BOEING 747/361	TO DETECT AND CORRECT CRACKING OF BULKHEAD WEB AT MULTIPLE SITES ALONG THE RADIAL LAP SLICE	FAA AD 2007-02-24	AS IN AD & SB	AS IN AD & SB
362	DGCA/BOEING 747/362	TO DETECT AND CORRECT CRACKING OF THE CREASE BEAM AND ADJACENT STRUCTURE	FAA AD 2007-01-09	AS IN AD & SB	AS IN AD & SB
363	DGCA/BOEING 747/363	TO PREVENT LEAKAGE OF FUEL OR FUEL VAPORS INTO AREAS WHERE IGNITION SOURCE MAY BE PRESENT	FAA AD 2007-07-03	AS IN AD & SB	AS IN AD & SB
364	DGCA/BOEING 747/364	TO DETECT AND CORRECT CRACKING OF THE AFT TENSION TIE CHANNELS	FAA AD 2007-16-19	AS IN AD & SB	AS IN AD & SB
365	DGCA/BOEING 747/365	TO DETECT AND CORRECT CRACKING IN THE FAIL-SAFE INTERLAYER OF CERTAIN NO.2 AND NO.3 GLASS WINDOWS	FAA AD 2007-15-10	AS IN AD & SB	AS IN AD & SB
366	DGCA/BOEING 747/366	TO DETECT AND CORRECT FATIGUE CRACKING OF THE FUSELAGE SKIN NEAR STRINGER 5	FAA AD 2007-15-07	AS IN AD & SB	AS IN AD & SB
367	DGCA/BOEING 747/367	TO DETECT AND CORRECT CRACK IN THE FUSELAGE SKIN AT THE CUTOUT OF THE BULK CARGO DOOR LIGHT	FAA AD 2007-17-13	AS IN AD & SB	AS IN AD & SB
368	DGCA/BOEING 747/368	TO PREVENT ARCHING INSIDE THE FUEL TANKS IN THE EVENT OF LIGHTNING STRIKE OR HIGH-POWERED SHORT CIRCUIT	FAA AD 2007-20-01	AS IN AD & SB	AS IN AD & SB

369	DGCA/BOEING 747/369	TO PREVENT INSUFFICIENT ELECTRICAL BONDING WHICH COULD RESULT IN A POTENTIAL OF IGNITION SOURCES INSIDE THE FUEL TANKS	FAA AD 2007-21-13	AS IN AD & SB	AS IN AD & SB
370	DGCA/BOEING 747/370	TO DETECT AND CORRECT CRACKING IN THE STA 1241 BULKHEAD FITTINGS	FAA AD 2007-25-17	AS IN AD & SB	AS IN AD & SB
371	DGCA/BOEING 747/371	TO PREVENT WATER CONTAMINATION OF THE ELECTRICAL/ELECTRONIC UNITS	FAA AD 2007-26-03	AS IN AD & SB	AS IN AD & SB
372	DGCA/BOEING 747/372	TO PREVENT THE IN LINE FLOW INDICATORS OF THE PASSENGER OXYGEN MASKS FROM FRACTURING AND SEPRATING	FAA AD 2007-26-06	AS IN AD & SB	AS IN AD & SB
373	DGCA/BOEING 747/373	reserved for ssad		AS IN AD & SB	AS IN AD & SB
374	DGCA/BOEING 747/374	TO DETECT AND CORRECT MIGRATED HINGE PINS AND DAMAGE FLIPPER DOORS	FAA AD 2007-26-07	AS IN AD & SB	AS IN AD & SB
375	DGCA/BOEING 747/375	TO PREVENT FAILURE OF AN EVACUATION SYSTEM	FAA AD 2008-03-05	AS IN AD & SB	AS IN AD & SB
376	DGCA/BOEING 747/376	TO DETECT AND CORRECT CRACKS AND BROKEN FASTNERS OF THE INBOARD AND OUTBOARD NACELLE STRUTS	FAA AD 2008-05-03	AS IN AD & SB	AS IN AD & SB
377	DGCA/BOEING 747/377	TO PREVENT A FRACTURED TRUNNION FORK ASSEMBLY WHICH COULD RESULT IN THE COLLAPSE OF A WING LANDING GEAR ON THE GROUND AND POSSIBLE DAMAGE TO HYD EQUIPMENT.	FAA AD 2008 - 09 -05	AS IN AD & SB	AS IN AD & SB
378	DGCA/BOEING 747/378	TO PREVENT CRACKING IN THE FUESLAGE SKIN WHICH COULD RESULT IN RAPID DECOMPRESSION & LOSS OF STRUCTURAL INTEGRITY OF THE AIRPLANE	FAA AD 2009 - 04 -16	AS IN AD & SB	AS IN AD & SB
379	DGCA/BOEING 747/379	TO PREVENT THE POTENTIAL FOR IGNITION SOURCES INSIDE FUEL TANKS CAUSED BY LATENT FAILURES, ALTERATIONS, REPAIRS, OR MAINTENANCE ACTIONS	FAA AD 2008-10-07	AS IN AD & SB	AS IN AD & SB
380	DGCA/BOEING 747/380	TO ENSURE FASTNERS IN THE SHEAR CLIP OF THE STUB FRAME TO AUXILIARY SILL JOINTS (LEFT AND RIGHT SIDES) ARE INSTALLED	FAA AD 2008-09-20	AS IN AD & SB	AS IN AD & SB
381	DGCA/BOEING 747/381	TO DETECT AND CORRECT BROKEN OR MISSIN FASTNERS IN THE HINGE SEGMENTS WITH A SINGLE FASTNER ROW WHICH COULD LEAD TO OPENING OF THE CARGO DOOR DURING FLIGHT	FAA AD 2008-14-08	AS IN AD & SB	AS IN AD & SB
382	DGCA/BOEING 747/382	TO DETECT AND CORRECT FUESLAGE SKIN CRACKS AT FASTENER LOCATIONS ALONG THE SKIN TO STRINGER ATTACHMENTS	FAA AD 2008-17-17	AS IN AD & SB	AS IN AD & SB
383	DGCA/BOEING 747/383	TO PREVENT LIGHTNING INDUCED ELECTRICAL ENERGY FROM ENTERING A RESERVE FUEL TANK THROUGH THE REFUEL VALVE,WHICH COULD RESULT IN A FUEL TANK EXPLOSION & CONSEQUENT LOSS OF AIRPLANE	FAA AD 2008-18 - 09	AS IN AD & SB	AS IN AD & SB
384	DGCA/BOEING 747/384	TO DETECT & CORRECT FATIGUE CRACKING OF THE LOWER FORWARD CORNER REVEAL OF THE NUMBER 3 MEDs WHICH COULD LEAD TO THE DOOR ESCAPE SLIDE DEPARTING THE AIRPLANE WHEN THE DOOR IS OPENED AND THE SLIDE IS DEPLOYED	FAA AD 2008-18 - 07	AS IN AD & SB	AS IN AD & SB
385	DGCA/BOEING 747/385	HAZARDS POSED BY SKEWING AND FAILED FLAPS	FAA AD 2008-23 - 10	AS IN AD & SB	AS IN AD & SB

386	DGCA/BOEING 747/386	TO ENSURE THAT INSULATION BLANKETS CONSTRUCTED OF AN -26 ARE REMOVED FROM THE FUSELAGE.SUCH INSULATION BRACKETS COULD IGNITE & PROPOGATE A FIRE THAT IS THE RESULT OF ELECTRICAL ARCING OR SPARKING	FAA AD 2008 -23 - 09	AS IN AD & SB	AS IN AD & SB
387	DGCA/BOEING 747/387	TO DETECT & CORRECT CRACKS OR FRACTURES OF THE PRIMARY STRUCTURAL & FAIL SAFE LOAD PATHS OF THE INBOARD & OUTBOARD TRAILING EDGE FLAPS	FAA AD 2008 -25 - 07	AS IN AD & SB	AS IN AD & SB
388	DGCA/BOEING 747/388	TO DETECT & CORRECT CRACKS IN THE UPPER LOBE DOUBLERS	FAA AD 2009 - 09 - 08	AS IN AD & SB	AS IN AD & SB
389	DGCA/BOEING 747/389	TO DETECT & CORRECT FATIGUE CRACKING OF THE LONGERON,WHICH CAN PROPOGATE & CAUSE DAMAGE TO THE ADJACENT HORIZONTAL STABILIZER PIVOT BULKHEAD.	FAA AD 2009 -12 - 08	AS IN AD & SB	AS IN AD & SB
390	DGCA/BOEING 747/390	TO DETECT & CORRECT FRAME CRACKS,WHICH COULD RESULT IN CRACKING OF THE ADJACENT FUSELAGE SKIN & CONSEQUENT RAPID DECOMPRESSION OF THE AIRPLANE	FAA AD 2009 -19 - 05	AS IN AD & SB	AS IN AD & SB
391	DGCA/BOEING 747/391	TO PREVENT A DECREASE OF THE AERODYNAMIC CONTROLLABILITY OF THE AIRPLANE WHICH COULD ADVERSELY AFFECT THE AIRPLANES CONTINUED SAFE FLIGHT & LANDING	FAA AD 2009 -20 - 12	AS IN AD & SB	AS IN AD & SB
392	DGCA/BOEING 747/392	TO PREVENT INJURIES TO PERSONS & DAMAGE TO THE AIRPLANE AND EQUIPMENT	FAA AD 2009 -22 - 08	AS IN AD & SB	AS IN AD & SB
393	DGCA/BOEING 747/393	TO DETECT & CORRECT MISSING FASTENERS AT THE STRINGER TO STRINGER CLIP JOINTS WHICH COULD RESULT IN SHEAR TIE & SKIN CRACKS & RAPID IN FLIGHT DECOMPRESSION OF THE AIRPLANE	FAA AD 2009 -24-17	AS IN AD & SB	AS IN AD & SB
394	DGCA/BOEING 747/394	TO DETECT & CORRECT FATIGUE CRACKING AT MULTIPLE ADJACENT LOCATIONS IN THE SUBJECT AREAS,WHICH COULD CONNECT TO FORM LARGE CRACKS AND RESULT IN REDUCED STRUCTURAL INTEGRITY LEADING TO RAPID DECOMPRESSION AND CONSEQUENT LOSS OF CONTROL OF THE AIRPLANE	FAA AD 2010 -05-03	AS IN AD & SB	AS IN AD & SB
395	DGCA/BOEING 747/395	TO PREVENT THE MALFUNCTION OF ONE OR MORE ELECTRICAL & ELECTRONIC UNITS IN THE MAIN EQUIPMENT CENTER,WHICH COULD ADVERSELY AFFECT THE AIRPLANES CONTINUED SAFE FLIGHT	FAA AD 2010-05-11	AS IN AD & SB	AS IN AD & SB
396	DGCA/BOEING 747/396	TO DETECT & CORRECT SUCH CRACKING WHICH COULD GROW & RESULT IN A SEVERED INTERCOSTAL .	FAA AD 2010-14-17	AS IN AD & SB	AS IN AD & SB
397	DGCA/BOEING 747/397	TO DETECT & CORRECT CRACKS & FRACTURES OF THE NACELLE STRUT FRONT SPAR CHORD ASSEMBLY	FAA AD 2010-14-09	AS IN AD & SB	AS IN AD & SB
398	DGCA/BOEING 747/398	TO PREVENT DAMAGE TO THE FUEL PUMPS CAUSED BY ELECTRICAL ARCING THAT COULD INTRODUCE AN IGNITION SOURCE IN THE FUEL TANK WHICH IN COMBINATION WITH FLAMMABLE FUEL VAPOURS COULD RESULT IN A FUEL EXPLOSION	FAA AD 2010-13 -12	AS IN AD & SB	AS IN AD & SB
399	DGCA/BOEING 747/399	TO PREVENT A POTENTIAL ELECTRICAL ARC FROM IGNITING THE BMS 8-39 POLYURETHANE FOAM INSULATION ON THE DUCT ASSEMBLIES OF THE ECS	FAA AD 2010-14-01	AS IN AD & SB	AS IN AD & SB
400	DGCA/BOEING 747/400	TO PREVENT INADVERTENT ELECTRICAL CURRENT,WHICH CAN CAUSE THE LOW PRESSURE FLEX HOSES OF THE CREW OXYGEN SYSTEM TO MELT OR BURN RESULTING IN OXYGEN SYSTEM LEAKAGE	FAA AD 2010-16-05	AS IN AD & SB	AS IN AD & SB

401	DGCA/BOEING 747/401	AIR TRANSPORT ASSOCIATION (ATA) OF AMERICA CODE 52:DOORS	FAA AD 2009 -19 - 06	AS IN AD & SB	AS IN AD & SB
402	DGCA/BOEING 747/402	AIR TRANSPORT ASSOCIATION (ATA) OF AMERICA CODE 28:FUEL	FAA AD 2010-24-13	AS IN AD & SB	AS IN AD & SB