

DGCANO	SUBJECT	REFERENCE	COMPLIANCE	APPLICABILITY
DGCA/CF34 Series/1	TO PREVENT UNCOMMANDED ENGINE ACCELERATIONS	FAA AD 99-23-26R1	'AS IN AD'	'AS IN AD'
DGCA/CF34 Series/2 (R3)	TO PREVENT AN UNCONTAINED FAILURE OF THE FAN DISK,WHICH COULD RESULT IN DAMAGE TO THE AIRPLANE	FAA AD 2009-26-09	'AS IN AD'	'AS IN AD'
DGCA/CF34 Series/3	TO PREVENT NO 5 BEARING ROTATING AIR SEAL FAILURES AND POSSIBLE UNCONTAINED ENGINE FAILURE	FAA AD 2001-12-06	'AS IN AD'	'AS IN AD'
DGCA/CF34 Series/4	TO PREVENT CRITICAL LIFE LIMITED ROTATING ENGINE PART FAILURE	FAA AD 2002-05-02	'AS IN AD'	'AS IN AD'
DGCA/CF34 Series/5 (R1)	TO PREVENT B-SUMPS SCAVENGE SCREEN BLOCKING DUE TO COKING	FAA AD 2003-05-10 R1	'AS IN AD'	'AS IN AD'
DGCA/CF34 Series/6(R1)	CANCELLED VIDE LETTER NO 9-766/03 AI(2) DATED 03/10/05			
DGCA/CF34 Series /7(R3)	TO PREVENT LOW CYCLE FATIGUE (LCF) FAILURE OF STAGE 5 LPT DISKS AND STAGE 6 LPT DISKS, WHICH COULD LEAD TO UNCONTAINED ENGINE FAILURE.	FAA AD 2006-04-12 (FAA AD 2004-15-03R1 IS SUPERSEDED)	'AS IN AD'	'AS IN AD'
DGCA/CF34 Series/8	TO PREVENT LCF CRACKS AND FAILURE OF THE HPC FORWARD SPOOL	FAA AD 2004-26-02	'AS IN AD'	'AS IN AD'
DGCA/CF34 Series/9	TO PREVENT DUAL-CHANNEL ELECTRICAL SIGNAL FAULTS IN THE VG MASTER ACTUATOR	FAA AD 2005-07-09	'AS IN AD'	'AS IN AD'
DGCA/CF34 Series/10	TO IMPOSE LIFE LIMITS ON CERTAIN HIGH PRESSURE TURBINE (HPT) ROTATING COMPONENTS	FAA AD 2005-18-16	'AS IN AD'	'AS IN AD'
DGCA/CF34 Series/11R1	TO PREVENT AN UNCONTAINED FAILURE OF THE FAN DISK,WHICH COULD RESULT IN DAMAGE TO THE AIRPLANE	FAA AD 2009-26-09	'AS IN AD'	'AS IN AD'
DGCA/CF34 Series/12	TO PREVENT ENGINE IN-FLIGHT SHUTDOWN DUE TO MFP MALFUNCTIONS	FAA AD 2007-12-15	'AS IN AD'	'AS IN AD'
DGCA/CF34 Series/13	TO PREVENT ENGINE FROM FAILING TO SHUTDOWN DURING AN OVERSPEED	FAA AD 2008-05-01	'AS IN AD'	'AS IN AD'
DGCA/CF34 Series/14R1	TO PREVENT LOSS OF THRUST CONTROL OF THE AIRPLANE	FAA AD 2009 -24-06	'AS IN AD'	'AS IN AD'
DGCA/CF34 Series/15	TO PREVENT THE INABILITY TO RESTART BOTH ENGINES AFTER FLAMEOUT DUE TO EXCESSIVE FRICTION OF THE 4 STEP SEAL WHICH COULD RESULT IN SUBSEQUENT FORCED LANDING OF THE AIRPLANE	FAA AD 2009-07-12	'AS IN AD'	'AS IN AD'
DGCA/CF34 Series/16R1	TO PREVENT FAILURE OF CERTAIN PART NUMBER (P/N) & SERIAL NUMBER (SN) FAN BLADES & AFT ACTUATOR HEAD HOSES, WHICH COULD RESULT IN AN UNDERCOWL FIRE & SUBSEQUENT DAMAGE TO THE AIRPLANE	FAA AD 2010 - 01- 04	'AS IN AD'	'AS IN AD'
DGCA/CF34 Series/17	TO PREVENT HIGH - CYCLE FATIGUE CRACKING OF THE FAN ROTOR SPINNER SUPPORT ATTACHMENT LUGS, LEADING TO SEPARATION OF THE FAN ROTOR SPINNER ASSEMBLY, UNCONTAINED FAILURE OF THE ENGINE, AND DAMAGE TO THE AIRPLANE.	FAA AD 2011-18-02	'AS IN AD'	'AS IN AD'