

DGCA NO	SUBJECT	REFERENCE	COMPLIANCE	APPLICABILITY
DGCA/CFM 56/1	TO PREVENT FAILURE OF NO. 3 BEARING	DGAC AD 89-181(B)R2	AS IN AD	AS IN AD
DGCA/CFM 56/2	TO PREVENT FAN BLADE FAILURE	FAA AD 96-25-11	AS IN AD	AS IN AD
DGCA/CFM 56/3	LIFE LIMITS, LOW PRESSURE TURBINE	DGAC AD 97-010(B)R1	AS IN AD	AS IN AB
DGCA/CFM 56/4	TO PREVENT A LOW CYCLE FATIGUE (LCF) FAILURE OF THE FAN DISK	FAA AD 97-08-01	AS IN AD	AS IN AD
DGCA/CFM 56/5	TO PREVENT A FAILURE OF A STAGE I DISK OF HIGH PRESSURE COMPRESSOR ROTOR (HPCR) STAGE 1-2 SPOOL	FAA AD 97-06-01	AS IN AD	AS IN AD
DGCA/CFM 56/6	TO PREVENT AN AIR TURBINE ENGINE STARTER FAILURE, WHICH COULD RESULT IN DAMAGE TO THE ENGINE ELECTRICAL HARNESS.	FAA AD 97-05-13	AS IN AD	AS IN AD
DGCA/CFM 56/7	ACCESSORY GEAR BOX AND TRANSFER GEAR BOX	DGAC AD 97-354 (B) R4	AS IN AD	AS IN AD
DGCA/CFM 56/8(R4)	FAN DISK INSPECTION	DGAC AD 97-298 R4	AS IN AD	AS IN AD
DGCA/CFM 56/9	HPT DISK INSPECTION	DGAC AD 97-327(B)	AS IN AD	AS IN AD
DGCA/CFM 56/10	ACCESSORY GEAR BOX (AGB) AND TRANSFER GEAR BOX (TGB)	DGAC AD 98-096 (B)	AS IN AD	AS IN AD
DGCA/CFM 56/11	STAGE 1-2 SPOOL HPC AND # 3 BEARING AFT STATIONARY	DGAC AD 98-080(B) R1	AS IN AD	AS IN AD
DGCA/CFM 56/12	ACCESSORY GEAR BOX (AFB)	DGAC AD T98-198(B) R1	AS IN AD	AS IN AD
DGCA/CFM 56/13	TO PREVENT AN UNCOMMANDED ENGINE ACCELERATION EVENT , OR INFLIGHT SHUTDOWN	DGAC AD T98-350 (B) R2	AS IN AD	AS IN AD
DGCA/CFM 56/14	ACCESSORY GEAR BOX	DGAC AD 98-259(B) R1	AS IN AD	AS IN AD
DGCA/CFM 56/15	FUEL FILTER COVER ATTACHMENT PARTS	DGAC AD 1999-245(B)R1	AS IN AD	AS IN AD
DGCA/CFM 56/16	CANCELLED IN VIEW OF DGCA/CFM 56/17 VIDE LETTER NO 9-719/02 DATED 04/10/05			
DGCA/CFM 56/17	TO DETECT CRACKS IN THE BOLT HOLES OF HPT FRONT ROTATING AIRSEALS	FAA AD 2000-05-22	AS IN AD	AS IN AD
DGCA/CFM 56/18	TO PREVENT CRITICAL LIFE-LIMITED ROTATING ENGINE PART FAILURE	FAA AD 2002-13-03	AS IN AD	AS IN AD
DGCA/CFM 56/19	TO PREVENT FUEL LEAKAGE FROM BETWEEN THE FUEL PUMP FILTER COVER & GEAR HOUSING	FAA AD 2000-15-01	AS IN AD	AS IN AD
DGCA/CFM 56/20	TO PREVENT AIR LEAKAGE FROM INCORRECTLY TORQUED FITTINGS OF THE PS3 PRESSURE LINE	FAA AD 2001-02-12	AS IN AD	AS IN AD
DGCA/CFM 56/21	CANCELLED IN VIEW OF DGCA/CFM56/08 (R4) VIDE LETTER NO 9-719/02 DATED 27/05/05			
DGCA/CFM 56/22	No. 4 BEARING (ATA 72)	DGAC AD 2001-240 (B)	AS IN AD	AS IN AD
DGCA/CFM 56/23	2ND AND 3RD STAGE LPT NOZZLE SEGMENTS	DGAC AD 2002-470(B)	AS IN AD	AS IN AD
DGCA/CFM 56/24	ENGINE FUEL AND CONTROL- REPLACEMENT OF MAIN FUEL PUMPS EQUIPPED WITH SOLID BRONZE BEARINGS WITH PUMPS EQUIPPED WITH BI-METAL, ALUMINIUM/ BRONZE BEARINGS	DGAC AD F 2004-095	AS IN AD	AS IN AD
DGCA/CFM 56/25	TO PREVENT UNCONTAINED FAILURE OF AIR TURBINE STARTERS	FAA AD 2005-10-05	AS IN AD	AS IN AD
DGCA/CFM 56/26	TO PREVENT UNEXPECTED DETERIOATION OF CRITICAL ROTATING ENGINE PARTS DUE TO HIGHER THEN DESIRED ENGINE OPERATION	FAA AD 2007-03-15	AS IN AD	AS IN AD
DGCA/CFM 56/27	TO PREVENT LOSS OF ENGINE THRUST	FAA AD 2006-26-01	AS IN AD	AS IN AD
DGCA/CFM 56/28	TIME LIMITS ---- LOW PRESSURE TURBINE REAR FRAME ----- LIFE REDUCTION	EASA AD 2007- 0104 / - 0221	AS IN AD	AS IN AD
DGCA/CFM 56/29	TO PREVENT LPT CONICAL SUPPORTS FROM REMAINING IN SERVICE BEYOND THEIR CERTIFIED CYCLIC LIFE LIMITS	FAA AD 2007- 24 -04	AS IN AD	AS IN AD

DGCA/CFM 56/30	TO PREVENT FAILURE OF THE TRF FROM LOW CYCLE FATIGUE CRACKS	FAA AD 2008-03-09	AS IN AD	AS IN AD
DGCA/CFM 56/31	TO PREVENT HPC STALLS WHICH COULD PREVENT CONTINUED SAFE FLIGHT OR LANDING	FAA AD 2009 - 01- 01	AS IN AD	AS IN AD
DGCA/CFM 56/32	ENGINE ----- FAN BLADES WITH 25 DEGREE MID SPAN SHROUDS ----- INSPECTIONS	EASA AD 2009 - 0036	AS IN AD	AS IN AD
DGCA/CFM 56/33R1	ENGINE --- HIGH PRESSURE COMPRESSOR (HPC) --- INSPECTION/REPLACEMENT	EASA AD 2009 - 0088R1 CORRECTED ON 17TH AUG 2010	AS IN AD	AS IN AD
DGCA/CFM 56/34	TIME LIMITS --- LOW PRESSURE TURBINE REAR FRAME --- LIFE LIMIT/MANDATORY INSPECTION	EASA AD 2009 - 0110	AS IN AD	AS IN AD
DGCA/CFM 56/35	TO PREVENT CRACKING OF THE HPC 4-9 SPOOL,WHICH COULD RESULT IN POSSIBLE UNCONTAINED FAILURE OF THE SPOOL & DAMAGE TO THE AIRPLANE	FAA AD 2009 - 11 - 02	AS IN AD	AS IN AD
DGCA/CFM 56/36	ENGINE FUEL & CONTROL --- HYDROMECHANICAL UNIT (HMU) --- INSPECTION / REPLACEMENT	EASA AD 2009 - 0154 -E	AS IN AD	AS IN AD
DGCA/CFM 56/37	ENGINE - LPT ROTOR / STATOR ASSEMBLY - REPLACEMENT	EASA AD 2009 - 0270	AS IN AD	AS IN AD
DGCA/CFM 56/38	TO PREVENT FAILURE OF THE LP TURBINE REAR FRAME FROM LOW CYCLE FATIGUE CRACKS	FAA AD 2010-01-05	AS IN AD	AS IN AD
DGCA/CFM 56/39	TO PREVENT FAILURE OF MULTIPLE FAN BLADES	FAA AD 2010-12-03	AS IN AD	AS IN AD
DGCA/CFM 56/40	TO PREVENT UNCONTAINED FAILURE OF THE STAGE THREE LPT DISK & DAMAGE TO THE AIRPLANE	FAA AD 2010 -13-09	AS IN AD	AS IN AD