

DGCA NO	SUBJECT	REFERENCE AD	COMPLIANCE	APPLICABILITY
DGCA/CONT IO 550/1	TO PREVENT EXHAUST RAM FAILURE	FAA AD 98-01-08	AS IN AD	AS IN AD
DGCA/CONT IO 550/2	TO PREVENT CRANKSHAFT FAILURE DUE TO NO 2 AND NO.5 CHECK CRACKS	FAA AD 99-09-17	AS IN AD	AS IN AD
DGCA/CONT IO 550/3	TO PREVENT CRANKSHAFT CONNECTING ROD JOINED FRACTURE	FAA AD 2000-23-21	AS IN AD	AS IN AD
DGCA/CONT IO 550/4	TO PREVENT TURBINE ROTOR FROM SEPARATING FROM THE SHAFT OF TURBOCHARGER DUE TO A MACHINING DEFECT IN THE TURBOCHARGER COMPRESSOR	FAA AD 2007-16-10	AS IN AD	AS IN AD
DGCA/CONT IO 550/5	TO PREVENT SEPARATION OF THE CYLINDER HEAD WHICH COULD RESULT IN IMMEDIATE LOSS OF ENGINE POWER	FAA AD 2009-16-03	AS IN AD	AS IN AD
DGCA/CONT IO 550/6R1	TO PREVENT LOSS OF ENGINE POWER DUE TO CRACKS IN THE CYLINDER HEAD, POSSIBLE ENGINE FAILURE, & FIRE IN ENGINE COMPARTMENT	FAA AD 2009 - 19 - 07 CORRECTED 7TH OCT 2009	AS IN AD	AS IN AD
DGCA/CONT IO 550/7	TO PREVENT EXCESSIVE HYDRAULIC LIFTER WEAR WHICH CAN RESULT IN LOSS OF ENGINE POWER & LOSS OF CONTROL OF THE AIRPLANE	FAA AD 2010-11-04	AS IN AD	AS IN AD