

DGCA NO	SUBJECT	REFERENCE AD	COMPLIANCE	APPLICABILITY
DGCA/CONT. O-470/1	REPLACEMENT OF PISTON PIN ASSEMBLY, OIL PUMP GEAR ASSEMBLY AND EXHAUST VALVE	FAA AD 56-06-01	AS IN AD	AS IN AD
DGCA/CONT. O-470/2	REPLACEMENT OF EXHAUST VALVES	FAA AD 63-15-01	AS IN AD	AS IN AD
DGCA/CONT. O-470/3	INSPECTION OF CARBURETTOR BOWL	FAA AD 73-13-04	AS IN AD	AS IN AD
DGCA/CONT. O-470/4	INSPECTION OF OIL PRESSURE RELIEF VALVE	FAA AD 79-05-09	AS IN AD	AS IN AD
DGCA/CONT. O-470/5	INSPECTION OF ENGINE AND CYLINDER ASSEMBLIES	FAA AD 85-08-02	AS IN AD	AS IN AD
DGCA/CONT. O-470/6	REPLACEMENT OF OIL FILTER P/Nos 649309 AND 649310 WITH A SERVICEABLE PART	FAA AD 88-03-06	AS IN AD	AS IN AD
DGCA/CONT. O-470/7	REPLACEMENT OF AIR METERING STOP PIN P/N 62-226 WITH AIR METERING STOP PIN P/N 62-F1	FAA AD 89-04-02	AS IN AD	AS IN AD
DGCA/CONT. O-470/8	REPLACEMENT OF CHAMPION OIL FILTERS P/N CH 48108 AND CH 48109	FAA AD 91-19-03	AS IN AD	AS IN AD
DGCA/CONT. O-470/9	INSPECTION TO CYLINDER VALVE RETAINER KEY TO THE ROTO COIL AND ITS REPAIR/REPLACEMENT	PRIORITY FAA ADS 93-10-02 AND TCM MANADATORY SB 93-12 DT. 12/5/93	AS IN AD & MSB	AS IN AD& MSB
DGCA/CONT. O-470/10	INSTALLATION OF THE CORRECT OIL PICK UP TUBE AND, IF NECESSARY, ITS REMOVAL AND REPLACEMENT.	FAA AD 93-08-17 AND TCM MANADATORY SB M91-10 R1 DT. NOV.27 1991	AS IN AD & SB	AS IN AD & SB
DGCA/CONT. O-470/11	TO PREVENT DETONATION DUE TO LOW OCTAIN, CAUSING ENGINE DAMAGE	FAA AD 95-21-15	AS IN AD	AS IN AD
DGCA/CONT. O-470/12	TO PREVENT EXTREME SIDE LOADING OF THE PISTON	FAA AD 97-21-02	AS IN AD & SB	AS IN AD & SB
DGCA/CONT. O-470/13	TO PREVENT EXHAUST ROLLER ROCHER AIM FAILURE	PRIORITY AD 98-01-08	AS IN AD & SB	AS IN AD & SB
DGCA/CONT. O-470/14	TO PREVENT CRANKSHAFT FAILURE DUE TO CRANKSHAFT CHEEKNAKS	FAA AD 99-19-01	AS IN AD & SB	AS IN AD & SB
DGCA/CONT. O-470/15	TO PREVENT CRANKSHAFT CONNECTTING ROD JOURNAL FRACTURE	FAA AD 2000-23-21	AS IN AD & SB	AS IN AD & SB
DGCA/CONT. O-470/16R1	TO PREVENT LOSS OF ENGINE POWER DUE TO CRACKS IN THE CYLINDER HEAD,POSSIBLE ENGINE FAILURE,& FIRE IN ENGINE COMPARTMENT	FAA AD 2009 - 19 - 07 CORRECTED 7TH OCT 2009	AS IN AD & SB	AS IN AD & SB
DGCA/CONT. O-470/17R1	TO PREVENT EXCESSIVE HYDRAULIC LIFTER WEAR WHICH CAN RESULT IN LOSS OF ENGINE POWER & LOSS OF CONTROL OF THE AIRPLANE	FAA AD 2010-11-04	AS IN AD & SB	AS IN AD & SB