

DGCA NO	SUBJECT	REFERENCE	COMPLIANCE	APPLICABILITY
DGCA/JT8D/1	INSPECTION OF FIRST STAGE COMPRESSOR ROTOR BLADES FOR TOOL MARKS.	F.A.A. A.D. NO. 70-26-05	AS IN A.D.	AS IN A.D.
DGCA/JT8D/2	REPLACEMENT OF EXISTING FUEL NOZZLE NUTS BY FUEL NOZZLE NUTS TO P/N. 672062.	P & W ALERT S.B.NO.3535 REV.1 DT.4/11/71	FORTHWITH	AS IN S.B.
DGCA/JT8D/3	FIRST STAGE FAN BLADE ROOT FAILURES DUE TO FREQUENCY FRACTURE.	P&W ALERT S.B.NO. 3580 REV.2 DT.17/03/72.	AS IN S.B.	AS IN S.B.
DGCA/JT8D/4	INCORPORATION OF NO.4 BEARING SEAL AIR FILTER.	P&W S.B.NO. 3178 REV.2 DT.19/8/71	DURING OVERHAUL	AS IN S.B.
DGCA/JT8D/5	TO PREVENT FAILURE OF THE FUEL NOZZLE SUPPORT ASSEMBLIES FROM FATIGUE CRACKING IN THE BASE FILLET.	F.A.A. A.D.NO. 71-25-07	AS IN A.D.	AS IN A.D.
DGCA/JT8D/6	SEVENTH STAGE COMPRESSOR DISC INSPECTION.	P&W ALERT S.B.NO. 4024 REV.11	AS IN S.B.	AS IN S.B.
DGCA/JT8D/7	INSPECTION OF FUEL MANIFOLD 'B' NUTS.	F.A.A. A.D.NO. 75-5-6 AND P&W ALERT S.B.NO. 4389 REV.11.	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
DGCA/JT8D/8	INSPECTION/REPLACEMENT OF ALL FOURTH STAGE STATOR VANES NOT FAA APPROVED FOR THAT ENGINE MODEL.	F.A.A. A.D.NO. 77-07-11	AS IN A.D.	AS IN A.D.
DGCA/JT8D/9	REWORK OF COMBUSTION CHAMBER PIN, BUSHINGS & LOCKS.	P&W S.B.NO. 4421	DURING O/H & OR NEXT SHOP VISIT OF THE ENGINE WHEN THE AFFECTED AREA IS OPENED UP FOR ANY REASON.	APPLICABLE TO ALL JT8D ENGINES.
DGCA/JT8D/10	DELETED VIDE DGCA LETTER NO. 9-471/86-AI(2) DT.30-06-87			
DGCA/JT8D/11	INSPECTION OF SECOND STAGE FAN BLADE PIN RETENTION RIVETS.	F.A.A. A.D.NO. 76-12-06 AND P&W ALERT S.B.NO. 4577 REV.1	WITHIN 400 FLIGHT HRS. FROM THE DATE OF ISSUE OF THIS LETTER.	AS IN S.B.
DGCA/JT8D/12	REMOVAL FROM SERVICE OF NO.4 1/2 BEARING P/N. 707007 WITH SERIAL NO. 800 AND ABOVE.	F.A.A. A.D.NO. 77-16-12 AND P&W ALERT S.B.NO. 4639	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
DGCA/JT8D/13	RETIREMENT LIFE OF 8TH STAGE COMPRESSOR DISK P/N. 496908.	F.A.A. A.D.NO. 76-24-01	AS IN A.D.	AS IN A.D.

DGCA/JT8D/14	INSPECTION FOR TIEROD HOLE CRACKING - HPC DISCS.	F.A.A. A.D.NO. 81-08-02 REV.2 AND P&W ALERT S.B.NO. 4723 REV.12.	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
DGCA/JT8D/15	ENGINE - FIRST STAGE FAN HUB BLADE SLOTS INSPECTION.	F.A.A. A.D.NO. 78-17-02 REV.2 AND P&W ALERT S.B.NO. 4841 REV.6	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
DGCA/JT8D/16	INSPECTION OF FUEL NOZZLE METERING SET HEAT SHIELD ELECTRON BEAM SEAM WELDS FOR FRACTURE.	P&W ALERT S.B.NO. 4871 REV.6	AS IN S.B.	AS IN S.B.
DGCA/JT8D/17	INSPECTION OF THIRD STAGE TURBINE BLADE FOR PROPER POSITION.	F.A.A. A.D.NO. 78-23-12 AND P&W S.B.NO. 4592 REV.1	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
DGCA/JT8D/18	INSPECTION OF 8TH STAGE COMPRESSOR.	F.A.A. A.D.NO. T80-15-51 AND P&W ALERT S.B.NO. 5154 REV.3	AS IN A.D. AND S.B.	ALL P&W JT8D ENGINES
DGCA/JT8D/19	INSPECTION AND REPAIR OF SECOND STAGE FAN BLADES.	P&W LETTER NO. JT8/72-33/PST:JFT/2-11-19-2 85 DT.19/11/1982.	AS IN P&W LETTER	AS IN P&W LETTER
DGCA/JT8D/20	CANCELLED IN VIEW OF FAA AD 85-19-51 REV.1			
DGCA/JT8D/21	TO PREVENT UNCONTAINED RUPTURE OF THE SECOND STAGE LPT DISK .	F.A.A. A.D.NO. 85-19-51 REV.1 AND P&W ALERT S.B.NOS. 5541 REV.2 AND 5510 REV.1.	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
DGCA/JT8D/22	INSPECTION OF COMBUSTION CHAMBER ASSEMBLY.	F.A.A. A.D.NO. 86-09-02 REV.1 AND P&W ALERT S.B.NO. 5639 REV.5.	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
DGCA/JT8D/23	SPACER ASSEMBLY COMPRESSOR ROTOR - REPLACEMENT OF REMOVABLE SLEEVE CONFIGURATIONS WITH INTEGRAL CONFIGURATIONS.	F.A.A. A.D.NO. 86-08-04 REV.1 AND P&W ALERT S.B.NO. 5649 REV.3.	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
DGCA/JT8D/24	CANCELLED.	CANCELLED.	CANCELLED.	CANCELLED.
DGCA/JT8D/25R1	TO PREVENT UNCONTAINED FAILURE OF 2ND STAGE FAN BLADES WHICH COULD RESULT IN DAMAGE TO THE AIRPLANE	FAA AD 2009 - 24 -01	AS IN A.D. AND S.B.	AS IN A.D. AND S.B.
DGCA/JT8D/26	REMOVAL OF NO.4 AND 1/2 BEARING P/N 707007 WITH SL. NO. BELOW 800 AND P/N	P & W ALERT S.B. NO. 5822	AS IN S.B.	AS IN S.B.

	735017 FROM SERVICE.	REV. 1.		
DGCA/JT8D/27	INCORPORATION OF LPC 1ST STAGE BLADE RETAINING PLATE WITH NEW RETAINING PLATE P/N 803996.	FAA AD 89-25-11 AND P & W ALERT S.B.NO. 5841	AS IN S.B.	AS IN S.B.
DGCA/JT8D/28	INSPECTION OF COMBUSTION CHAMBER CASE OUTER WALL FOR CIRCUMFERENTIAL CRACKS.	FAA AD 94-25-07 AND ASB 6230.	AS IN AD SB AND SUBSEQUENT REVISIONS TO ASB.	AS IN AD AND ASB.
DGCA/JT8D/29	INSPECTION OF 3RD & 4TH STAGE TURBINE ROTOR BLADES.	FAA AD 97-19-14 & FAA AD 94-20-08	AS IN AD & ASB	AS IN AD& ASB
DGCA/JT8D/30	PREVENTION OF NO. 4 1/2 BEARING FAILURE.	FAA AD 91-24-14	AS IN AD	AS IN AD
DGCA/JT8D/31	INITIAL AND REPETITIVE INSPECTIONS OF CERTAIN FRONT COMPRESSOR FAN HUBS AND SHOTPEENING THE FORWARD AND AFT RIB TO WEB RADIUS.	FAA AD 93-14-14 AND PW ASB NO. 6104 R2 DT. JUNE 18 '1993	AS IN AD & SB.	AS IN AD & SB.
DGCA/JT8D/32	REPETITIVE INSPECTION OF NO. 7 FUEL NOZZLE AND SUPPORT ASSEMBLY AND REPLACEMENT.	FAA AD 94-14-16 AND PW ASB A6153 R1	AS IN AD AND ASB.	AS IN AD AND ASB.
DGCA/JT8D/33	CANCELLED IN VIEW OF DGCA/JT8D/47			
DGCA/JT8D/34	TO PREVENT DAMAGE TO A/C RESULTING FROM ENGINE DEBRIS FOLLOWING A LPT BLADE OR SHAFT FAILURE.	FAA AD 94-20-09 & ASB NO. 6039 R3.	AS IN AD & ASB.	AS IN AD & ASB.
DGCA/JT8D/35	CANCELLED.			
DGCA/JT8D/36	TO PREVENT AN UNCONTAINED (HPC) DISK FAILURE & DAMAGE TO A/C.	FAA 95-26-03 & PW ASB NO. A6626.	AS IN AD & ASB.	AS IN AD & ASB.
DGCA/JT8D/37	TO PREVENT THE INITIATION AND PROPAGATION OF A FATIGUE CRACK FRACTURE OF THE FAN HUB, CONTAINED ENGINE FAILURE AND DAMAGE TO THE AIRCRAFT.	FAA AD 96-15-06	AS IN AD.	AS IN AD.
DGCA/JT8D/38	TO PREVENT CCOC FLANGE CRACKS THAT COULD RESULT IN UNCONTAINED ENGINE FAILURE, INFLIGHT ENGINE SHUTDOWN, ENGINE COWL RELEASE AND AIRFRAME DAMAGE.	FAA AD 96-23-14 ASB 6228 ASB 6202(IF NOT CERTIFIED IN ACCORDANCE WITH 6228)	AS IN AD ASB AND SUBSEQUENT REVISIONS TO ASB.	AS IN AD SB AND SUBSEQUENT REVISIONS TO ASB.
DGCA/JT8D/39	ENGINE-CASE ASSEMBLY, COMBUSTION CHAMBER OUTER REPLACEMENT OF	SB 5966	AT THE NEXT SHOP VISIT AFTER 1-1-98	AS IN SB
DGCA/JT8D/40	TO PREVENT 4TH STAGE LPT BLADE FAILURE DUE TO HUB CRACKING WHICH CAN RESULT IN UNCONTAINED ENGINE FAILURE & DAMAGE TO THE AIRCRAFT	FAA AD 97-19-12	AS IN AD	AS IN AD
DGCA/JT8D/41	TO PREVENT UNCONTAINED ENGINE FAILURE, INFLIGHT ENGINE SHUTDOWN, ENGINE COWL RELEASE AND AIRFRAME DAMAGE	FAA AD 98-04-39	AS IN AD	AS IN AD
DGCA/JT8D/42	TO PREVENT THE FAILURE OF HIGH PRESSURE COMPRESSOR DISK	FAA AD 99-01-08	AS IN AD	AS IN AD
DGCA/JT8D/43(R 2)	TO PREVENT CRITICAL LIFE-LIMITED ROTATING ENGINE PART FAILURE	FAA AD 2011 -03-01	AS IN AD	AS IN AD
DGCA/JT8D/44	TO PREVENT LOSS OF ENGINE THROTTLE CONTROL	FAA AD 2000-22-06	AS IN AD	AS IN AD

DGCA/JT8D/45	TO PREVENT RUPTURE OF 2ND STAGE COMPRESSOR DISC CAUSED BY MACHINING DAMAGE	FAA AD 2001-15-18	AS IN AD	AS IN AD
DGCA/JT8D/46	TO PREVENT 1ST AND 2ND STAGE FAN SECTION SEPARATION FROM THE LPC	FAA AD 2003-05-07	AS IN AD	AS IN AD
DGCA/JT8D/47	TO PREVENT FRACTURE OF THE 7TH, 8TH, 9TH, 10TH, 11TH & 12TH STAGE HIGH PRESSURE COMPRESSOR (HPC) DISKS	FAA AD 2003-12-07	AS IN AD	AS IN AD
DGCA/JT8D/48	TO PREVENT FAILURE OF CRITICAL LIFE-LIMITED ROTATING ENGINE PARTS THAT HAVE RETURNED TO SERVICE WITH CRACKS, CORROSION PITTING, OR DIMENSION OUTSIDE OF MANUAL LIMITS	FAA AD 2005-17-16	AS IN AD	AS IN AD
DGCA/JT8D/49(R1)	TO PREVENT A RUPTURE OF HIGH PRESSURE COMPRESSOR (HPC) FRONT HUB OR HPC DISK	FAA AD 2006-17-07R1	AS IN AD	AS IN AD
DGCA/JT8D/50	TO PREVENT FAN HUB FAILURE DUE TO TIEROD,COUNTERWEIGHT, OR BUSHED HOLE CRACKING,WHICH COULD RESULT IN AN UNCONTAINED ENGINE FAILURE	FAA AD 97-17-04R1	AS IN AD	AS IN AD
DGCA/JT8D/51	TO PREVENT HIGH CYCLE FATIGUE CRACKING AT THE BLADE ROOT WHICH COULD RESULT IN UNCONTAINED FAILURES OF FIRST STAGE FAN BLADES AND DAMAGE TO THE AIRPLANE	FAA AD 2010-21-17	AS IN AD	AS IN AD
DGCA/JT8D/52	TO PREVENT CRITICAL LIFE LIMITED ROTATING ENGINE PART FAILURE,WHICH COULD RESULT IN AN UNCONTAINED ENGINE FAILURE AND DAMAGE TO THE AIRPLANE	FAA AD 2011-04-04	AS IN AD	AS IN AD
DGCA/JT8D/53	TO PREVENT TURBINE BLADE FAILURES THAT COULD RESULT IN UNCONTAINED ENGINE DEBRIS & DAMAGE TO THE AIRPLANE	FAA AD 2011-07-02	AS IN AD	AS IN AD