

DGCANO	SUBJECT	REFERENCE	COMPLIANCE	APPLICABILITY
DGCA/PW4000/1	MODIFICATION OF AIRCRAFT ENGINE IDLE SYSTEM WIRING.	FAA AD 93-08-16 AND ASB 747-73A2055	AS IN AD & ASB	AS IN AD & ASB
DGCA/PW4000/2	TO PREVENT AFT CASCADE SUPPORT FRAME ASSEMBLY FAILURE DUE TO CRACKS WHICH CAN RESULT IN TRUST REVERSER HARDWARE LIBERATION AND REJECTION FROM THE AIRCRAFT DURING THRUST REVERSER OPERATION.	FAA AD 96-13-08	AS IN AD.	AS IN AD.
DGCA/PW4000/3	TO PREVENT 2ND STAGE HPT ROTATING AIRSEAL CRACKING.	FAA AD 98-19-03	AS IN AD	AS IN AD
DGCA/PW4000/4	TO PREVENT HIGH PRESSURE COMPRESSOR (HPC) SURGE.	FAA AD 98-23-08	AS IN AD.	AS IN AD
DGCA/PW4000/5(R1)	TO PREVENT CRITICAL LIFE LIMITED ROTATING ENGINE PART FAILURE	FAA AD 2005-25-09	AS IN AD	AS IN AD
DGCA/PW4000/6	TO PREVENT AN UNCONTAINED DISK FAILURE	FAA AD 99-15-01	AS IN AD	AS IN AD
DGCA/PW4000/7	CANCELLED IN VIEW OF DGCA/PW4000/10			
DGCA/PW4000/8	CANCELLED IN VIEW OF DGCA/PW4000/10			
DGCA/PW4000/9	TO REDUCE STRESSES THAT COULD LEAD TO FATIGUE CRACKING OF THE LPT 4TH STAGE AIR SEAL	FAA AD 2001-05-07	AS IN AD	AS IN AD
DGCA/PW4000/10	TO PREVENT ENGINE TAKE OFF POWER LOSSES DUE TO HPC SURGES	FAA AD 2003-19-15	AS IN AD	AS IN AD
DGCA/PW4000/11	TO PREVENT 2ND STAGE HPT AIR SEAL ASSEMBLY FAILURE	FAA AD 2001-20-02	AS IN AD	AS IN AD
DGCA/PW4000/12	TO DETECT PREMATURE CRACKING OF THE HIGH PRESSURE COMPRESSOR (HPC) FRONT DRUM ROTOR	FAA AD 2002-12-15	AS IN AD	AS IN AD
DGCA/PW4000/13	TO PREVENT 15th STAGE HPC DISK FAILURES FROM CRACKS WHICH COULD RESULT IN AN UNCONTAINED ENGINE FAILURE	FAA AD 2002-09-01	AS IN AD	AS IN AD
DGCA/PW4000/14	TO PREVENT THERMAL DISTRESSED HPT ASSEMBLY HARDWARE FROM REMAINING IN SERVICE	FAA AD 2003-26-09	AS IN AD	AS IN AD
DGCA/PW4000/15	TO PREVENT FRACTURING AND RUPTURING OF THE HPC REAR CASE	FAA AD 2003-13-11	AS IN AD	AS IN AD
DGCA/PW4000/16(R1)	REPLACING THE HIGH PRESSURE COMPRESSOR (HPC) EXIT BRUSH SEAL ASSEMBLY	FAA AD 2007-04-26	AS IN AD	AS IN AD
DGCA/PW4000/17	TO PREVENT UNCONTAINED ENGINE FAILURE	FAA AD 2006-22-13	AS IN AD	AS IN AD
DGCA/PW4000/18	TO PREVENT AN UNCONTAINED ENGINE FAILURE DUE TO LOW CYCLE FATIGUE	FAA AD 2007-26-20	AS IN AD	AS IN AD
DGCA/PW4000/19	TO PREVENT UNCOMMANDED IN FLIGHT ENGINE SHUTDOWNS WHICH COULD RESULT IN LOSS OF THRUST & PREVENT CONTINUED SAFE FLIGHT	FAA AD 2008 - 24 13	AS IN AD	AS IN AD
DGCA/PW4000/20	TO PREVENT AN UNCONTAINED FAILURE OF THE FRONT TURBINE HUB,RESULTING IN AN IN FLIGHT ENGINE SHUTDOWN & POSSIBLE DAMAGE TO THE AIRPLANE	FAA AD 2009 - 04 - 02	AS IN AD	AS IN AD
DGCA/PW4000/21	TO DETECT CRACKS IN THE LOCKING AND LOADING SLOTS IN THE HPC DRUM ROTOR DISK ASSEMBLIES,WHICH COULD RESULT IN RUPTURE OF THE HPC DRUM ROTOR DISK ASSEMBLY	FAA AD 2010-18-13	AS IN AD	AS IN AD
DGCA/PW4000/22	TO PREVENT CRACKING OF NO 3 BEARING OIL PRESSURE TUBES,WHICH COULD RESULT IN INTERNAL OIL FIRE,FAILURE OF THE HIGH PRESSURE TURBINE DISKS,UNCONTAINED ENGINE FAILURE,AND DAMAGE TO THE AIRPLANE	FAA AD 2010-24-09	AS IN AD	AS IN AD
DGCA/PW4000/23	TO PREVENT FAILURE OF THE HPC 10TH STAGE DISK,UNCONTAINED ENGINE FAILURE,AND DAMAGE TO THE AIRPLANE	FAA AD 2010-24-14	AS IN AD	AS IN AD
DGCA/PW4000/24	TO PREVENT CRACKS FROM PROPAGATING INTO THE BOLT HOLES OF THE 15TH STAGE HPC DISK, WHICH COULD RESULT IN A FAILURE OF THE 15TH STAGE HPC DISK	FAA AD 2011-14-07	AS IN AD	AS IN AD