DGCA NO	SUBJECT	REFERENCE AD	COMPLIANCE	APPLICABILITY
DGCA / TFE-731 / 1	TO PREVENT SEPARATION OF THE ROD AND POTENTIAL MASSIVE ENGINE FAILURE	FAA AD 80-16-04 AND AIR RESEARCH SB TFE 731-72-3105 OR 3106	AS IN AD & SB	AS IN AD & SB
DGCA / TFE-731 / 2	TO PREVENT SEPARATION OF THE ENGINE REAR MOUNT FROM THE DUCT	FAA AD 81-24-08 AND GARRETT S.B. TFE 731-72-3159	AS IN AD & SB	AS IN AD & SB
DGCA / TFE-731 / 3	PREVENTION OF FAILURE OF HIGH PRESSUE COMPRESSOR IMPELLER FAN AND COMPRESSOR ROTOR DISCS SERVICE LIFE LIMTS TO PARTS AS IN PARA C OF A.D.	FAA AD 82-23-03 R1 AND GARRETT S.B. TFE 731-72-3239 RWK	AS IN AD & SB	AS IN AD & SB
DGCA / TFE-731 / 4	ENGINE VISUAL INSPECTION - TURBINE INTERSTAGE TRANSITION DUCT	FAA AD 86-03-01 AND GARRETT S.B. TFE 731-72-3159 R5	AS IN AD & SBS	AS IN AD & SBS
DGCA / TFE-731 / 5	CANCELLED	IN VIEW OF DGCA/TFE 731/14	AS IN AD & ASB	AS IN AD & ASB
DGCA / TFE-731 / 6	INSPECTION OF HIGH PRESSUE TURBINE ROTOR (HPTR) DISC	FAA AD 88-18-03 AND GARRETT ASB TFE 731-A72-3376	AS IN AD & ASB	AS IN AD & ASB
DGCA / TFE-731 / 7	REPLACEMENT OF HPTR DISCS WITH A SERVICEABLE PARTS TO PREVENT AN UNCONTAINED ENGINE FAILURE	FAA AD 90-03-04 AND GARRETT ASB TFE 731-A72-3388	AS IN AD & ASB	AS IN AD & ASB
DGCA / TFE-731 / 8	TO PREVENT OVERTEMPERATURE AND DAMAGE TO TURBINE COMPONENTS	FAA AD 90-24-06 AND GARRETT S.B. TFE 731-A77-3020	AS IN AD & SB	AS IN AD & SB
DGCA / TFE-731 / 9	EDDY CURRENT OR FLUORESCENT PENETRANT INSPECTION FOR CRACKS - FAN ROTOR DISCS, AND THEIR RE-INSPECTION	FAA AD 96-05-03 AND BLAD 1996-035/AB	AS IN AD & ASB	AS IN AD & ASB
DGCA / TFE-731 / 10	TO PREVENT VNCONTAINED FAILURE OF THE FIRST STAGE LP TURBINE DISK	FAA AD 94-07-03	AS IN AD	AS IN AD
DGCA / TFE-731 / 11	EDDY CURRENT INSPECTION OF FAN DISKS AND THEIR REPLACEMENT IF NECESSARY	FAA AD 96-04-01	AS IN AD & ASB	AS IN AD & ASB
DGCA / TFE-731 / 12	TO PREVENT CRACKED FUEL TUBES AND SUBSEQUENT LINKAGE OF FUEL ON OR AROUND ELECTRICAL COMPONENTS	FAA AD 98-17-01 R1 AND GARRETT SB TFE 731-73-3118 R1	AS IN AD & SB	AS IN AD & SB
DGCA / TFE-731 / 13	TO PREVENT LOW PRESSURE TURBINE (LPT) DISK WEB SEPARATION	FAA AD 95-07-02	AS IN AD	AS IN AD
DGCA / TFE-731 / 14	TO PREVENT FAILURE OF FAN ROTOR DISC	FAA AD 2001-23-09	AS IN AD	AS IN AD
DGCA / TFE-731 / 15	TO PREVENT FATIGUE CRACKING AND SUBSEQUENT UNCONTAINED FAILURE OF FIRST STAGE LPT SEAL PLATE	FAA AD 97-04-03	AS IN AD	AS IN AD
DGCA / TFE-731 / 16	TO PREVENT FUEL SPRAYING ON AND AROUND ELECTRICAL COMPONENT DUE TO A FUEL CRACKED LINE	FAA AD 98-10-15	AS IN AD & SB	AS IN AD & SB
DGCA / TFE-731 / 17	ENGINE FUEL & CONTROL - FUEL FLOW METER TUBE ASSEMBLY	FAA AD 99-07-19	AS IN AD & SB	AS IN AD & SB
DGCA / TFE-731 / 18	TO PREVENT FAILURE OF THE HPC IMPELLER DUE TO FATIGUE CRACKING	FAA AD 2000-15-09	AS IN AD & SB	AS IN AD & SB
DGCA / TFE-731 / 19	TO PREVENT ADDITIONAL UNCONTAINED FAILURE OF THE LPT STAGE 1 DISK	FAA AD 2005-05-15	AS IN AD & SB	AS IN AD & SB
DGCA / TFE-731 / 20	TO PREVENT LP TURBINE ROTOR DISK SEPERATION	FAA AD 2005-13-07	AS IN AD & SB	AS IN AD & SB
DGCA / TFE-731 / 21	TO PREVENT UNCONTAINED FAILURE OF THE HP TURBINE ROTOR ASSEMBLY	FAA AD 2008-02-19	AS IN AD & SB	AS IN AD & SB
DGCA / TFE-731 / 22	TO PREVENT UNCONTAINMENT OF TURBINE BLADES & FRAGMENTS OF THE LPT NOZZLE ASSEMBLY,WHICH COULD RESULT IN DAMAGE TO THE AIRPLANE	FAA AD 2008 -19 -12	AS IN AD & SB	AS IN AD & SB
DGCA / TFE-731 / 23	TO PREVENT AN UNCONTAINED FAILURE OF A SECOND STAGE LPCR DISC AND / OR A THIRD STAGE LPCR DISC DUE TO CRACKS IN THE BORE WHICH COULD RESULT IN DAMAGE	FAA AD 2010-06-11	AS IN AD & SB	AS IN AD & SB