

DGCA NO	SUBJECT	REFERENCE	COMPLIANCE	APPLICABILITY
DGCA/V2500/01	IAE V2500-A1 ENGINE REAR MOUNT BARREL NUTS	BVAD 90-079-009(B)R1	AS IN AD	AS IN AD
DGCA/V2500/02	TO PREVENT AN HPT DISK FRACTURE	FAA AD 99-25-03	AS IN AD.	AS IN AD.
DGCA/V2500/03	TO PREVENT FAN BLADE ROOTCRACKS	FAA AD 98-21-01	AS IN AD	AS IN AD
DGCA/V2500/04	CANCELLED IN VIEW OF DGCA/V2500/2			
DGCA/V2500/05	FORWARD THRUST LINKS REPLACEMENTS (ATA 71)	DGAC AD 1999-050-126(B)	AS IN AD	AS IN AD
DGCA/V2500/06	REMOVAL OF CERTAIN HPC STAGE 9-12 DRUMS PRIOR TO REACHING THE NEW REDUCED CYCLE LIFE LIMITS	FAA AD 99-04-03	AS IN AD	AS IN AD
DGCA/V2500/07(R1)	TO PREVENT CRITICAL LIFE LIMITED ROTATING ENGINE PART FAILURE	FAA AD 2004-12-08, ORG FAA AD 2000-12-05 IS SUPERSEDED	AS IN AD	AS IN AD
DGCA/V2500/08	TO PREVENT FAILURE OF HPT HARDWARE DUE TO IGNITION OF OIL TRAPPED BY THE HPT STAGE 1 INNER ROTATING AIRSEAL(TOBI- ID SEAL) WHICH COULD RESULT IN AN UNCONTAINED ENGINE FAILURE & DAMAGE TO AIRPLANE	FAA AD 99-13-01	AS IN AD	AS IN AD
DGCA/V2500/09	TO PREVENT POSSIBLE LOW CYCLE FATIGUE (LCF)FAILURE OF THE HPT STAGE1DISK	FAA AD 2000-22-07	AS IN AD	AS IN AD
DGCA/V2500/10(R1)	FUEL FILTER COVER- INSPECTION / MODIFICATION	EASA AD 2005-0012 (ORIGINAL DGAC AD 2003-355(B)R2 IS SUPERSEDED)	AS IN AD	AS IN AD
DGCA/V2500/11(R1)	TO PREVENT FAILURE OF THE NO 3 BEARING	FAA AD 2006-25-01 (FAA AD SUPERSEDES AD 2003-11-23)	AS IN AD	AS IN AD
DGCA/V2500/12	TO PREVENT UNCONTAINED FAILURE OF THE HPC STAGE 3 TO 8 DRUM, WHICH COULD RESULT IN DAMAGE TO THE AIRPLANE.	FAA AD 2010-20-07	AS IN AD	AS IN AD