

DGCA NO	SUBJECT	REFERENCE	COMPLIANCE	APPLICABILITY
DGCA/BELL 430/1	TO SECURE THE INSTALLATION OF THE FOUR RESOLVER CONTROL ARMS	TC AD CF 99-26	AS IN AD	AS IN AD
DGCA/BELL 430/2	CALIBRATION OF THE FUEL QUANTITY INDICATING SYSTEM	TC AD CF 2000-04	AS IN AD	AS IN AD
DGCA/BELL 430/3	TO PREVENT LOSS OF TORQUE OF THE VERTICAL FIN ATTACHMENT BOLTS	FAA AD 99-23-03	AS IN AD	AS IN AD
DGCA/BELL 430/4	ELECTRICAL SYSTEM IMPROVEMENTS	TC AD CF 2000-08	AS IN AD	AS IN AD
DGCA/BELL 430/5	ELECTRICAL SYSTEM GROUND FAULTS	TC AD CF 2000-32R2	AS IN AD	AS IN AD
DGCA/BELL 430/6	TO INSPECT THE ACTUATOR SUPPORT ATTACHMENT	TC AD CF 2000-29	'AS IN AD'	'AS IN AD'
DGCA/BELL 430/7	DRAIN PROVISION	TC AD CF 2001-22	'AS IN AD'	'AS IN AD'
DGCA/BELL 430/8	REMOVAL OF AIR TEMPERATURE RESTRICTION	TC AD CF 1997-23R3	'AS IN AD	'AS IN AD
DGCA/BELL 430/9	EMERGENCY BUS ISOLATION	TC AD CF 2002-27	'AS IN AD'	'AS IN AD'
DGCA/BELL 430/10	TAIL ROTOR BLADE CRACKED	TC AD CF 2004-21R3	'AS IN AD'	'AS IN AD'
DGCA/BELL 430/11	SERVO ACTUATOR BOLTS- REPLACEMENT AND MANUAL UPDATE	TC AD CF 2005-09	'AS IN AD'	'AS IN AD'
DGCA/BELL 430/12	POWER TURBINE RPM STEADY STATE OPERATION AVOIDANCE	TC AD CF 2005-25	'AS IN AD'	'AS IN AD'
DGCA/BELL 430/13(R1)	TAIL ROTOR COUNTERWEIGHT BELLCRANK	TC AD CF 2005-27(R1)	'AS IN AD'	'AS IN AD'
DGCA/BELL 430/14	VNE- INSTRUMENTATION CHANGE	TC AD CF 2005-30	'AS IN AD'	'AS IN AD'
DGCA/BELL 430/15	MAIN ROTOR OVERSPEED WHEN SWITCHING FADEC TO MANUAL MODE	TC AD CF 2005-33	'AS IN AD'	'AS IN AD'
DGCA/BELL 430/16	FUEL VALVE SWITCH FAILURE	TC AD CF 2006-03	'AS IN AD'	'AS IN AD'
DGCA/BELL 430/17	TAIL ROTOR PITCH CHANGE MECHANISM	TC AD CF 2007-04	'AS IN AD'	'AS IN AD'
DGCA/BELL 430/18	DISC ASSEMBLIES --- TAIL ROTOR DRIVESHAFT (THOMAS COUPLING)	TC AD CF 2007- 15	'AS IN AD'	'AS IN AD'
DGCA/BELL 430/19	BEARINGS IN FLIGHT CONTROLS	TC AD CF - 2009 -32	'AS IN AD'	'AS IN AD'
DGCA/BELL 430/20	TRANSMISSION PLANETARY PINION GEAR DAMAGE	TCAD CF -2010-11	'AS IN AD'	'AS IN AD'
DGCA/BELL 430/21R1	TO DETECT CORROSION OR A NON CONFORMING GRIND RELIEF ON THE OUTPUT PISTON ROD ASSEMBLY(PISTON ROD), TO PREVENT FAILURE OF THE PISTON ROD,FAILURE OF THE SERVO ACTUATOR,& SUBSEQUENT LOSS OF CONTROL OF HELICOPTER	FAA AD 2010-19-51	'AS IN AD'	'AS IN AD'
DGCA/BELL 430/22	TO PREVENT LOSS OF A BLADE TIP WEIGHT,LOSS OF A BLADE,AND SUBSEQUENT LOSS OF CONTROL OF THE HELICOPTER	FAA AD 2010-26-51	'AS IN AD'	'AS IN AD'