

DGCA NO	SUBJECT	REFERENCE AD	COMPLIANCE	APPLICABILITY
DGCA / BELL-47/1	INSPECTION OF FUEL PUMP ROCKER PIN	FAA AD 48-13-02	AS IN AD & SB	APPLICABLE ALL MODEL 47 HELICOPTER WITH FRANKLIN ENGINES
DGCA / BELL-47/2	ROTATION OF DAMPERS FOR CONTINUOUS OIL SUBMERSION OF STABILIZER BAR DAMPER REPLENSHING VALVE	FAA AD 48-19-01	AS IN AD & SB	APPLICABLE TO MODEL 47 D ONLY
DGCA / BELL-47/3	INSPECTION OF ATTACHING CLAMPS AND BOLTS ON THE VENTRAL FIN	FAA AD 49-34-02	AS IN AD & SB	APPLICABLE TO MODEL 47 DI ONLY
DGCA / BELL-47/4	INSPECTION & REPLACEMENT TAIL ROTOR DRIVE SHAFT	FAA AD 49-47-02	AS IN AD & SB	APPLICABLE TO MODEL 47 D & 47 D1 ONLY
DGCA / BELL-47/5	INSPECTION AND REPLACEMENT OF MAIN ROTOR HUB	FAA AD 49-52-02	AS IN AD & SB	APPLICABLE TO MODEL 47 D & 47 D1 ONLY
DGCA / BELL-47/6	REPLACEMENT OF VENTRAL FIN	FAA AD 50-06-01	FORTH WITH	APPLICABLE TO MODEL 47 D & 47 D1 ONLY
DGCA / BELL-47/7	ADJUSTMENT OF FAN BELT TENSION	FAA AD 50-38-02	FORTH WITH	APPLICABLE TO MODEL 47 D & 47 D1 ONLY
DGCA / BELL-47/8	TO PROVIDE ADDITIONAL STRENGTH AND REPLACEMENT OF CRACKOD BLADES	FAA AD 51-26-01	FORTH WITH	47 & 47 D1 ONLY APPLICABLE
DGCA / BELL-47/9	INSPECTION AND REPLACEMENT OF ENGINE MOUNT	FAA AD 51-27-01	FORTH WITH	APPLICABLE TO ALL 47 MODEL ENGINE MOUNTASSY P/N 47-612-111-1 OR-5
DGCA / BELL-47/10	REPLACEMENT OF TAIL ROTOR GEAR BOX P/N 47-645-028-1 OR 47-645-051-1	FAA AD 51-29-01	AS IN AD & SB	APPLICABLE TO MODEL 47D & 47 D1 ONLY
DGCA / BELL-47/11	TO PROVIDE LOCKS OF INCREASED TENSILE STRENGTH	FAA AD 52-01-01	FORTH WITH	APPLICABLE TO MODEL 47D & 47 D1 ONLY
DGCA / BELL-47/12	REPLACEMENT OF EQUALIZER HORN & DRAG BRACE FITTING INVOLVED IN ACCIDENT	FAA AD 52-01-02	FORTH WITH	47D ONLY
DGCA / BELL-47/13	REPLACEMENT OF EXISTING DAMPER LOWER & LINK ASSEMBLIES	FAA AD 52-01-03	FORTH WITH	47 D ONLY
DGCA / BELL-47/14	TO PROVIDE TAIL ROTOR BLADE STRIP RETAINING BOLT OF INCREASED SERVICE LIFE	FAA AD 52-01-05	FORTH WITH	47 D ONLY
DGCA / BELL-47/15	INSTALLATION OF SPOILER ON TAIL ROTOR BOOM	FA AD 52-01-06	FORTH WITH	ALL MODEL 47 WITH COVERED TAIL BOOM
DGCA / BELL-47/16	REWORK OF HUB PART NO.47-661-030-1 IN THE COOLING FAN	FAA AD 52-01-07	FORTH WITH	ALL MODEL 47 D WITH AFFECT HUB
DGCA / BELL-47/17	INSTALLATION OF SAFETY STRAP KIT P/N 47-612-132	FAA AD 52-05-03	FORTH WITH	ALL 47 MODEL WITH ENGINE MOUNT P/N 47-612-11-1 OR 5 47 D1 ONLY
DGCA / BELL-47/18	TO PROVIDE SIMPLIFIED LOADING CHART	FAA AD 52-07-01	FORTH WITH	47 D1 ONLY
DGCA / BELL-47/19	INSPECTION OF ENGINE RUBBER LORD MOUNTS FOR POSSIBLE SHIFTING	FAA AD 52-21-01	FORTH WITH	APPLICABILITY AS IN AD
DGCA / BELL-47/20	REPLACEMENT OF TAIL ROTOR GEAR BOX SHAFT P/N 47-645-077-1	FAA AD 52-28-03	AS IN AD	ALL 47 MODEL WITH AFFECTED PART
DGCA / BELL-47/21	INSPECTION & REWORK OF TAIL ROTOR PINION SHAFT FOR CRACKS	FAA AD 52-28-04	FORTH WITH	47 D & 47 D1 MODEL ONLY
DGCA / BELL-47/22	INSTALLATION OF SLEEVES TO IMPROVE METHOD OF MOUNTING TAIL ROTOR GEAR BOX	FAA AD 53-01-01	FORTH WITH	MODEL 47 D1 ONLY

DGCA / BELL-47/23	INSTALLATION OF STOP ASSEMBLES FOR DUAL CONTROL ANTI TORQUE PEDAL ASSY	FAA AD 54-24-01 BELL S.B. NO.97 REV.A DT. 03.12.54	FORTH WITH	TO ALL 43 D, 47 D1 & 47 G (WITH CERTAIN SR.NOS.)
DGCA / BELL-47/24	REPLACEMENT OF PITCH CONTROL LINK BEARING	FAA AD 55-05-01 BELL SB NO.97 REV.A DT. 03.12.54	FORTH WITH	ALL 47 D D, 47 D1 & 47G MODELS
DGCA / BELL-47/25	ADDITIONAL SUPPORT FOR NO.1 BEARING HANGER	FAA AD 55-14-01	FORTH WITH	ONLY TO 47 G MODEL BELOW SR. NO.1317
DGCA / BELL-47/26	INSPECTION OF TAIL ROTOR HUB IN BOARD FILLET AREA	FAA AD 56-16-01	FORTH WITH	ALL 47 MODELS WITH P/N 47-641-025-1 TAIL ROTOR HUBS
DGCA / BELL-47/27	INSPECTION OF THE BOLT HOLES IN THE CYCLIC BELL CRANK FOR ELONGATION	FAA-AD 55-25-01	FORTH WITH	APPLICABLE TO MODEL 47G BELOW SR.NO.1450 ONLY
DGCA / BELL-47/28	INSPECTION OF TAIL BOOM LONGERON AND VENTRAL FIN SUPPORT ANGLE BRACE	FAA AD 56-01-01	FORTH WITH	TO MODEL 47G ONLY
DGCA / BELL-47/29	INSPECTION OF TAIL ROTOR EXTENSION DRIVE SHAFT P/N 47-6-44-192-3	FAA AD 56-05-01	FORTH WITH	TO ALL MODELS 47 D1 & 47G(CERTAIN SR. NOS.)
DGCA / BELL-47/30	INSPECTION OF CONTROL RING HORN IN THE SWASH PLATE	FAA AD 56-15-02	AS IN AD & SB	ALL 47 MODELS EXCEPT CERTAIN SR.NO. AS MENTIONED IN AD
DGCA / BELL-47/31	REPLACEMENTS OF TAIL ROTOR HUB RETENTION BOLTS	FAA AD 56-20-03	AS IN AD & SB	APPLICABLE TO MODELS 47 D,47D1 & 47 G, 47 G2
DGCA / BELL-47/32	REPLACEMENT OF HYDRAULIC BOOST POWER CYLINDERS	FAA AD 56-21-01 & BELL SB NO.114 DT. 27-08-56	AS IN AD & SB	APPLICABLE TO ALL MODEL 47 G AND 47 G2 ONLY
DGCA / BELL-47/33	REPLACEMENT OF WASHERS IN TAIL ROTOR PITCH CHANGE ROD ASSEMBLY TO PREVENT LOSS OF BEARING	FAA AD 56-25-02	AS IN AD & SB	APPLICABLE TO MODELS 47 G ONLY
DGCA / BELL-47/34	INSPECTION OF TAIL ROTOR HUB EQUALIZER LINK ASSEMBLY	FAA AD 57-07-02	AS IN AD & SB	ALL 47 MODELS EXCEPT HOSE LISTED IN AD
DGCA / BELL-47/35	INSPECTION OF TAIL ROTOR DRIVE SHAFT P/NS AS DETAILED IN FAA AD 57-8-1	FAA AD 57-08-01	FORTH WITH	ALL 47 MODELS WITH AFFECTED P/N AS DETAILED IN AD
DGCA / BELL-47/36	INSPECTION AND REPLACEMENT OF PARKER CHECK VALVE IN OIL SYSTEM	FAA AD 57-08-02	FORTH WITH	MODELS 47 D & 47 D1 ONLY
DGCA / BELL-47/37	INSPECTION OF METAL TAIL ROTOR BLADES	FAA AD 57-12-04	FORTH WITH	ALL MODELS 47 D & 47 D1 AND 47G WITH METAL TAIL ROTOR BLADE
DGCA / BELL-47/38	INSPECTION OF TAIL ROTOR PITCH CHANGE SHAFT AND REPLACEMENT OF BEARING	FAA AD 58-23-01	AS IN AD & SB	ALL MODELS 47D, & 47D1, 47G & G1
DGCA / BELL-47/39	REWORKING OF ENGINE COOLING FAN INDEX PLATE	FAA AD 58-26-01	FORTH WITH	MODEL 47 G2 ONLY
DGCA / BELL-47/40	MODIFICATION OF ENGINE MIXTURE CONTROL	FAA AD 59-26-02	AS IN AD & SB	ALL MODELS 47 HELICOPTERS WITH MARVEL SCHEBLER CURB MOD. MAA-5AA
DGCA / BELL-47/41	NSPECTION OF METAL TAIL ROTOR BLADES P/N 47-642-102-1, -5M 7,9-17 AND 19	FAA AD 59-05-01	FORTHWITH	ALL MODEL 47 WITH AFFECTED BLADES
DGCA / BELL-47/42	INSPECTION OF SCISSOR LEVER PIVOT BOLT	FAA AD 60-10-01	FORTHWITH	TO MODEL 47D, 47D1 AND 47G
DGCA / BELL-47/43	INSPECTION OF MAIN ROTOR PITCH LINK ROD & BEARING	FAA AD 61-22-01	FORTHWITH	ALL MODEL 47 HELICOPTERS

DGCA / BELL-47/44	INSPECTION OF MAIN ROTOR MAST THRUST BEARING	FAA AD 62-23-02	FORTHWITH	ALL MODEL 47 HELICOPTERS WITH AFFECTED BEARING AS MENTIONED IN AD
DGCA / BELL-47/45	INSPECTION OF WOODEN TAIL ROTOR BLADE P/N 47-642-020-1	FAA AD 62-27-02 DT. 29-01-63	AS IN AD & SB	ALL MODEL 47 WITH WOODEN TAIL ROTOR BLADES
DGCA / BELL-47/46	REPLACEMENT OF CLUTCH OUT RACES WHOSE HARDNESS ARE NOT WITHIN LIMIT	FAA AD 65-04-01 DT. 17-02-65	AS IN AD & SB	ALL MODEL 47 WITH TRANSMISSION SR.NO.B-12-885 THROUGH B12-969
DGCA / BELL-47/47	INSPECTION AND REPLACEMENT OF TAIL ROTOR METAL BLADES P/N 47-642-102	FAA AD 70-10-08	AS IN AD & SB	ALL MODEL 47 WITH METAL ROTOR BLADES P/N 102 SB NO.43 REV REFERS THE SAME SUBJECT
DGCA / BELL-47/48	REPLACEMENT OF TAILROTOR DRIVE SHAFT	BELL SB NO.47-145 REV.C	AS IN SB	AS IN SB
DGCA / BELL-47/49	INSPECTION OF DRAG LINK EQUALIZER HORN P/N 47-120-167-1 FOR HARDNESS	FAA AD 70-08-02 FAA AD 70-11-04 BELL SB NO.144	AS IN AD & SB	AS IN AD & SB
DGCA / BELL-47/50	TAIL ROTOR DUPLEX PITCH CHANGE BEARING & DELTA HINGE BOLTS	FAA AD 70-14-01	AS IN AD & SB	47-D1,G, G2&1 - 1
DGCA / BELL-47/51	TAIL ROTOR HUB ASSEMBLY BOLT	FAA AD 70-14-02	AS IN AD & SB	47-D1,G, G2&1 - 1
DGCA / BELL-47/52	INSPECTION OF BOTH CONTROL TUBES FOR CRACKS BY DYE PENETRANT METHOD	FAA AD 72-02-05 DATED 22-02-72	AS IN AD & SB	AS IN AD & SB
DGCA / BELL-47/53	IMPOSITION OF LIFE LIMIT OF 400 HRS TO TAIL ROTOR YOKE P/N 47-641-057-9	FAA AD 73-19-2; FAA AD 76-12-02	AS IN AD & SB	APPLICABLE TO MODELS 47D;1, 47G, 47G-2
DGCA / BELL-47/54	INSPECTION OF MAIN ROTOR GRIP THREAD RELIEF FILLET	FAA AD 73-19-04	AS IN AD & SB	MODEL 47 SERIES HELICOPTER FITTED WITH MAIN ROTOR GRIPS P/N 47-120-135-1, -2, -3 OR 47-120-252-1, -3
DGCA / BELL-47/47	INSPECTION AND REPLACEMENT OF TAIL ROTOR METAL BLADED P/N 47-642-102	FAA AD 70-10-08	AS IN AD & SB	ALL MODEL 47 WITH METAL ROTOR BLADES P/N 102 SB NO.43 REV REFERS THE SAME SUBJECT
DGCA / BELL-47/48	REPLACEMENT OF TAIL ROTOR DRIVE SHAFT	BELL SB NO.47-145 REV.C	AS IN SB	AS IN SB
DGCA / BELL-47/49	INSPECTION OF DRAG LINK EQUALIZER HORN P/N 47-120-167-1 FOR HARDNESS	FAA AD 70-08-02 FAA AD 70-11-04 BELL SB NO.144	AS IN AD & SB	AS IN AD & SB
DGCA / BELL-47/50	TAIL ROTOR DUPLEX PITCH CHANGE BEARING & DELTA HINGE BOLTS	FAA AD 70-14-01	AS IN AD & SB	47-D1,G, G2&1 - 1
DGCA / BELL-47/51	TAIL ROTOR HUB ASSEMBLY BOLT	FAA AD 70-14-02	AS IN AD & SB	47-D1,G, G2&1 - 1
DGCA / BELL-47/52	INSPECTION OF BOTH CONTROL TUBES FOR CRACKS BY DYE PENETRANT METHOD	FAA AD 72-02-05 DATED 22-02-72	AS IN AD & SB	AS IN AD & SB
DGCA / BELL-47/53	IMPOSITION OF LIFE LIMIT OF 400 HRS TO TAIL ROTOR YOKE P/N 47-641-057-9	FAA AD 73-19-02; FAA AD 76-12-02	AS IN AD & SB	APPLICABLE TO MODELS 47 D;1, 47G, 47G-2
DGCA / BELL-47/54	INSPECTION OF MAIN ROTOR GRIP THREAD RELIEF FILLET	FAA AD 73-19-04	AS IN AD & SB	MODEL 47 SERIES HELICOPTERS FITTED WITH MAIN ROTOR GRIPS P/N 47-120-135-1,-2,-3 OR 47-120-252-1,-3
DGCA / BELL-47/55	CYCLIC & COLLECTIVE MAIN ROTOR CONTROL TUBE ASSEMBLY	FAA AD 74-07-01 DT.	AS IN AD & SB	MODEL 47 SERIES HELICOPTERS

	INSTALLED WITHIN 30 OF VERTICAL AXIS	30-5-74		
DGCA / BELL-47/56	INSPECTION OF DETECT POSSIBLE FAILURES OF A STABILIZER BAR TUBE & WEIGHT	FAA AD 74-08-02 DT. 14-06-74	AS IN AD & SB	AS IN AD & SB
DGCA / BELL-47/57	INSPECTION OF REWORK FOR CRACKS ON CENTRE FRAME ASSEMBLY	DGCA LETTER NO. 15/15/72-AI(I) DT. 26-03-75	AS STATED IN THE DGCA LETTER	MODEL BELL 47 G-5
DGCA / BELL-47/58	INSPECTION OF SINGLE ROW BALL BEARING P/N 47-641-146-1 IN THE TAIL ROTOR PITCH CONTROL CROSS-HEAD	FAA AD 75-18-06	AS IN AD & SB	AS IN AD & SB
DGCA / BELL-47/59	PERIODIC INSPECTION ON DAMPER CLAMP ASSAY P/N 47-140-134-1	D.G.C.A. LETTER NO.78-99/74-AI(I) DT.31-1075	WITH IMMEDIATE EFFECT	ALL BELL 47 HELICOPTER
DGCA / BELL-47/60	INSTALLATION OF IMPROVED TAIL ROTOR HUB BOLTS & SEATS	FAA AD 76-12-01	AS IN SB	AS IN SB
DGCA / BELL-47/61	CANCELLED ON 04-05-76			8 B 5
DGCA / BELL-47/62	MANDATORY RETIREMENT OF MAIN ROTOR BLADE GRIPS P/N 47-120-135-1, -2 & -3	BELL SB NO 76-3 DT. 15-3-76 REV. 13.9.76	EFFECTIVE DT.15-04-76	AS IN SB
DGCA / BELL-47/63	MANDATORY RETIREMENT OF MAIN ROTOR GIMBAL RINGS P/N 47-120-014-5,-6,-7 ETC.	BELL SB NO.47-76-4 DT.05-04-76	31-03-77	AS IN SB
DGCA / BELL-47/64	INSPECTION OF TAIL ROTOR GEAR BOX P/N 47-640-075-1 & -7	BELL SB NO. 47-77-01 DT. 14-02-77	AFTER RECEIPT OF THE SB OR 30-04-77 WHICHEVER IS EARLY	AS IN SB
DGCA / BELL-47/65	INSPECTION OF HOPPER UPPER SUPPORT BRACKET	FAA AD 78-03-01	AS IN AD & SB	AS IN AD & SB
DGCA / BELL-47/66	INSPECTION OF COUNTER WEIGHT ROD PT. 47-110	BELL SB NO.47-78	AS IN SB	AS IN SB
DGCA / BELL-47/67	INSTALLATION OF AN ADDITIONAL SAFETY WASHER, WITH THE TAIL ROTOR PITCH LINK	BELL SB NO.47-80-05 REV.A & FAA AD 80-21-09R1	AS IN SB	AS IN SB
DGCA / BELL-47/68	RETIREMENT LIFE OF TAIL ROTOR BLADE PT.NO. 47-642-102	FAA AD 80-10-04	31ST OCT. 1981 OR RECEIPT OF PARTS OR 300 HRS SERVICE LIFE, WHICH EVENT IS EARLIER	ALL BELL 47 HELICOPTER
DGCA / BELL-47/69	REPLACEMENT OF BEARING PT NO.47-620-605-001, WITH BEARING PT.NO.47-620-929-001	FAA AD 80-04-03 & BELL ALERT SB NO.47-80-06	AS IN SB	AS IN SB
DGCA / BELL-47/70	INSPECTION/REPLACEMENT OF INCORRECT AN /NAS STANDARD BOLTS INSTALLED IN CERTAIN FLIGHT CONTROL APPLICATIONS	BELL ALERT SB NO.47-83-08 FAA AD 85-02-06	AS IN AD & SB	AS IN AD & SB
DGCA / BELL-47/71	INSPECTION OF SERVO-CYLINDERS P/N 103650-3, 103654-5 ETC. AND 47-691-273-1	BELL ALERT SB 47-83-09	AS IN SB	AS IN SB
DGCA / BELL-47/72	REPLACEMENT OF TAIL ROTOR DRIVE SHAFT ASSEMBLY P/N 47-644-234-3	BELL HELICOPTER ALERT SB 47-84-10 (REV.A)	AS IN SB	AS IN SB
DGCA / BELL-47/73	REMOVAL FROM SERVICE TAIL ROTOR BLADE ASSEMBLIES	FAA AD 85-08-03	AS IN AD & SB	AS IN AD & SB

	SL.NO.47-642-117-1 SL NO.A-8 THROUGH A-721	BELL ALERT SB NO.47-84-11		
DGCA / BELL-47/74	TO PREVENT FAILURE OF TAIL ROTOR BLADE JOKE	FAA AD 85-15-04	AS IN AD & SB	AS IN AD & SB
DGCA / BELL-47/75	TO PREVENT FAILURE OF A MAIN ROTOR BLADE GRIP	FAA AD 2001-17-17	AS IN SB	AS IN SB
DGCA / BELL-47/76	ONE TIME INSPECTION OF THE MAIN ROTOR GRIPS FOR POTENTIALLY UNSAFE GRIPS	FAA AD 91-20-10	AS IN AD & SB	AS IN AD & SB
DGCA / BELL-47/77	TO PREVENT PRE MATURE DELEMINATION OR SEPARATIONS OF THE TAIL ROTOR BLADE TIP BLOCK	FAA AD 96-06-12	AS IN AD & SB	AS IN AD & SB
DGCA / BELL-47/78	TO PREVENT FAILURE OF THE STABILIZER BAR DAMPER LINK ASSEMBLY	FAA AD 98-07-09	AS IN AD & SB	AS IN AD & SB
DGCA / BELL-47/79	TO ESTABILISE A LIFE LIMIT TO PREVENT USING A SCISSORS ASSEMBLY OR WELD ASSEMBLY SCISSORS BRACKET PAST ITS LIFE LIMITS	FAA AD 2005-21-04	'AS IN AD'	'AS IN AD'
DGCA/ BELL-47/ 80	TO PREVENT A MAIN ROTOR BLADE SPAR CRACK	FAA AD 2009-12-01	'AS IN AD'	'AS IN AD'