

DGCA NO	SUBJECT	REFERENCE	COMPLIANCE	APPLICABILITY
DGCA/HILLER/01	TO PREVENT SEPARATION OF THE CONTROL ROTOR BLADE ASSEMBLY AND SUBSEQUENT LOSS OF CONTROL OF THE HELICOPTER	FAA AD 97-10-16	'AS IN AD & SUBSEQUENT REVISIONS TO SB'	'AS IN AD & SUBSEQUENT REVISIONS TO SB'
DGCA/HILLER/02	TO PREVENT CRACKS IN THE HEAD AREA OF THE OUTBOARD T-T BAR PIN	FAA AD 97-20-15	'AS IN AD & SUBSEQUENT REVISIONS TO SB'	'AS IN AD & SUBSEQUENT REVISIONS TO SB'
DGCA/HILLER/03	TO PREVENT SEPARATION OF THE CONTROL SYSTEM ATTACHMENTS AT PIVOTING POINTS	FAA AD 2000-24-21	'AS IN AD & SUBSEQUENT REVISIONS TO SB'	'AS IN AD & SUBSEQUENT REVISIONS TO SB'