

DGCA NO	SUBJECT	REFERENCE	COMPLIANCE	APPLICABILITY
DGCA/R44/1	TO PREVENT FAILURE OF THE CYCLIC CONTROL SYSTEM	FAA AD 94-26-10	'AS IN AD'	'AS IN AD'
DGCA/R44/2	TO PREVENT M/R STALL OR MAST BUMPING	FAA AD 95-04-13	'AS IN AD'	'AS IN AD'
DGCA/R44/3	TO PREVENT IN FLIGHT M/R SEPARATION	FAA AD 95-11-10	'AS IN AD'	'AS IN AD'
DGCA/R44/4	TO PREVENT BINDING IN THE CYCLIC CONTROL SYSTEM	FAA AD 95-09-07	'AS IN AD'	'AS IN AD'
DGCA/R44/5	TO MINIMISE THE POSSIBILITY OF THE PILOT MIS ENGAGEMENT OF THE M/R RPM	FAA AD 97-02-15	'AS IN AD'	'AS IN AD'
DGCA/R44/6	TO PREVENT LOOSENING OF THE BOLTS SECURING THE GFEAR TO THE GEAR CARRIER	FAA AD 96-18-22	'AS IN AD'	'AS IN AD'
DGCA/R44/7	TO DETECT EXCESSIVE WEAR ON THE LATERAL CYCLIC TRIM SPRING SHAFT	FAA AD 98-0412	'AS IN AD'	'AS IN AD'
DGCA/R44/8	TO PREVENT FAILURE OF EITHER 45DEG. OR 90DEG. AL. ELBOWS	FAA AD 98-05-10	'AS IN AD'	'AS IN AD'
DGCA/R44/9	TO PREVENT FAILURE OF M/R BLADE	FAA AD 98-22-16	'AS IN AD'	'AS IN AD'
DGCA/R44/10	TO RPEVENT USE OF ASSEMBLY THAT MAY CRACK RESULTING FAILURE	FAA AD 98-21-36	'AS IN AD'	'AS IN AD'
DGCA/R44/11	TO ALERT PILOTS OF THE POTENTIAL FOR THE SPRAG CLUTCH FAILING TO OVERRUN DURING AUTOROTATION	FAA AD 99-0718	'AS IN AD'	'AS IN AD'
DGCA/R44/12	TO PREVENT FUEL LEAKS FROM THE AUXILIARY FUELTANK SUMP DRAIN	FAA AD 99-13-11	'AS IN AD'	'AS IN AD'
DGCA/R44/13	TO PREVENT FAILURE OF YOKE ASSEMBLY	FAA AD 99-17-17	'AS IN AD'	'AS IN AD'
DGCA/R44/14	FLIGHT MANUAL	DGAC AD 2000-075(A)	'AS IN AD'	'AS IN AD'
DGCA/R44/15	TO PREVENT SPRAG CLUTCH FAILURE, LOSS OF MAIN ROTOR RPM DURING AUTOROTION	FAA AD 2000-08-04	'AS IN AD'	'AS IN AD'
DGCA/R44/16	TO PREVENT CONTACT BETWEEN THE WIRE HARNESS & THE FUEL LINE ASSEMBLY	FAA AD 2000-07-03	'AS IN AD'	'AS IN AD'
DGCA/R44/17	TO PREVENT A CRACK THROUGH A VERTICAL TO HORIZONTAL STABILIZER ATTACH CHANNE	FAA AD 2001-20-18	'AS IN AD'	'AS IN AD'
DGCA/R44/18	TO DETECT CORROSION OF A TAIL ROTOR PITCH CONTROL BEARING	FAA AD 2003-04-05	'AS IN AD'	'AS IN AD'
DGCA/R44/19	TO PREVENT FAILURE OF THE BOLT THAT SECURES PARTS OF THE MAIN ROTOR SWASHPLATE AND ATTACHES THE TAIL ROTOR BLADE	FAA AD 2003-24-51	'AS IN AD'	'AS IN AD'
DGCA/R44/20(R1)	JASC CODE 6210 - MAIN ROTOR BLADES	FAA AD 2011-12-10 SUPERSEDES FAA AD 2007-26-12	'AS IN AD'	'AS IN AD'
DGCA/R44/21	TO PREVENT THE TAIL ROTOR (T/R) CONTROL PEDAL BEARING BLOCK SUPPORT FROM BREAKING,WHICH CAN BIND THE T/R CONTROL PEDALS,RESULTING IN AREDUCTION OF YAW CONTROL & SUBSEQUENT LOSS OF CONTROL OF HELICOPTER	FAA AD 2010-24-03	'AS IN AD'	'AS IN AD'